# CERTIFICATION OF APPROVAL 

# Analysis And Design Of A Multi-Storey Reinforced Concrete Structure Using Staid Pro And Robot Millennium 

by
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## CERTIFICATION OF ORIGINALITY

This is to certify that I am responsible for the work submitted in this project, that the original work is my own except as specified in the references and acknowledgements, and that the original work contained herein have not been undertaken or done by unspecified sources or persons.

$\overline{\text { LEE TSE WENG }}$


#### Abstract

The study on comparative analysis and design of Reinforced Concrete Structures using Application Software available in UTP is presented in the Final Year Project. A Reinforced Concrete structure model, which is created with STAAD PRO and ROBOT MILLENNIUM are analyzed. To verify the effectiveness of these software, the Reinforced Concrete beams, columns and slabs are analyzed according to British Standard (BS) 8110. During the progress stage of the research, a few reinforced concrete structures examples have been analyzed and designed. These examples consist of 2 dimensional and 3 dimensional frame structures. It is observed in the analysis that, the operability and the result output has some slight difference. Geometrical and material modeling plays an important role in determining the accuracy of the results in the reinforced concrete analysis. The analysis result indicates that a study on local behavior and effects must be carried out to ensure better result. Later, the research will focus on the common results between the software, whereby certain degrees of variations will be compared with manual calculations. Finally, discussion will be made on the variations and recommendations will be suggested based on these analysis.


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## CHAPTER 1 INTRODUCTION

### 1.1 BACKGROUND

In Malaysia, concrete plays an important role as building material in construction works. Concrete is a strong durable material, which made up from mixture of cement, sand, aggregates, and water with specific ratio standard can be formed into varied shapes and sizes. Nowadays, there is a high demand in construction development, and there is a need to accelerate the design process. Therefore, design software is used to speed up the analysis, design, detailing of structures in the design office. Precise methods of analysis of such as threedimensional structures can effectively only are carried out using these design software. Thus, it can eliminate the tedious manual calculation works. However, the flood of analysis and design software in the market has aroused the question of the effectiveness in terms of analysis, design and detailing.

A lot of structural software is being used for design purposes in the market nowadays. In UTP, there are a few structural and design software purchased for the benefit of the structural engineering community in the university. However, up to this moment the software have not been implemented in any structural courses yet. It is important to verify the results obtained $b$ efore it $c$ an be implemented in the course. Many aspects must be considered from analysis to design points of view. For this reason, some structural software in UTP will be analyzed thoroughly. It is then verified with manual calculation.

### 1.2 RESEARCH OBJECTIVE

The purpose of the research is to perform analysis and design of multiple structures building according to British Standard (BS) 8110 using design software. Throughout the research, the usage for the STAAD PRO and ROBOT MILLENNIUM can be determined. This research will help to improve understanding of the analysis and design of Concrete Building(s) and individual elements, design processes, design philosophy, method and approach. From the obtained results, the detailing ability embedded in these software can be identified and compared.

### 1.3 SCOPE OF STUDY

This research involved numerical and theoretical analysis. These analysis are based on load-deflection, load- strain and cracking behavior of the reinforced concrete structure.

The scope of the studies can be divided into: the study of Reinforced Concrete related to the research, the study of operation and usability and verification of results from STAAD PRO and ROBOT MILLENNIUM design software. Results obtained were then were analyzed and discussed.

## CHAPTER 2 LITERATURE REVIEW

### 2.1 HISTORY OF REINFORCED CONCRETE

Concrete is a compound material made from sand, gravel and cement. The cement is a mixture of $v$ arious minerals which when mixed with water, hydrate and rapidly become hard binding the sand and gravel into a solid mass. The oldest known surviving concrete is to be found in the former Yugoslavia and was thought to have been laid in $5,600 \mathrm{BC}$ using red lime as the cement.

The first major concrete users were the Egyptians in around 2,500 BC and the Romans from 300 BC . It is from the Roman words 'caementum' meaning a rough stone or chipping and 'concretus' meaning grown together or compounded, that we have obtained the names for these two now common materials. ${ }^{1}$

In 1830, a publication entitled, "The Encyclopedia of Cottage, Farm and Village Architecture" suggested that a 1 attice of iron rods could be embedded in concrete to form a roof. Eighteen years later, a French lawyer created a sensation by building a boat from a frame of iron rods covered by a fine concrete which he exhibited at the Paris Exhibition of 1855 . Steel reinforced concrete was now born. ${ }^{2}$

It is not only fire resistance that is improved by the inclusion of steel in the concrete matrix. Concrete, although excellent in compression, performs poorly when in tension or flexure. By introducing a network of connected steel bars, the strength under tension is dramatically increased allowing long, unsupported runs of concrete to be produced. Concrete also protects the steel, both physically and chemically. ${ }^{2}$

The Romans made many developments in concrete technology including the use of lime and Pozzolana concretes were used for nearly two millennium before the next major development occurred. In 1824 when Joseph Aspdin of Leeds took out a patent for the manufacture of Portland cement, so named because of its close resemblance to Portland stone. Aspdin's cement, made from a mixture of clay and limestone, which had been crushed and fired in a kiln, was an immediate success. Although many developments have since been made, the basic ingredients and processes of manufacture are the same today. ${ }^{2}$

This history clearly describes the importance of Reinforced Concrete as a building tool in construction material. Therefore, this fundamental process can be identified for future development application.

### 2.2 DEVELOPMENT OF BRITISH STANDARD CODES

The design procedure done for this research is according to British Standard (BS) Codes. For this reason, it is important to identify the guidelines information of this Code. This part of BS 8110: Code of Practice for the Structural use of Concrete has been prepared to replace CP110: Part 1:1972. This code covers the fields of CP110 and encompasses the structural use of reinforced and prestressed concrete both cast in situ and precast.

Although there are no major changes in principle from the previous edition, the text has largely been rewritten with alterations in the order and arrangement of topics.

The redrafting and alterations have been made in the light of experience of the practical convenience in using CP110. They have also been made to meet criticism of
engineers preferring the form of CP114. In this respect sections two to five have been rewritten with shorter clauses, avoiding as much as possible lengthy paragraphs dealing with the matters that could be broken down into separate subclauses, to make specific references easier to understand. From this development, consideration had been given to include the load factor method, which had been introduced into CP114 in 1957.3

BS 8110 is divided into 3 parts:

Part 1: Code of Practice for Design and Construction. This section covers the design objectives and general recommendations, design and detailing for reinforced concrete and prestressed concrete. This section also provides important information on concrete: materials, specification and construction. Besides that, the specification and workmanship were also explained thoroughly.

Part 2: Code of Practice for Special Circumstances. This Part gives guidance on ultimate limit state calculations and the derivation of partial factors of safety, serviceability calculations with emphasis on deflections under loading and on cracking

Part 3: Design Charts for Singly Reinforced Beams, Doubly Reinforced and Rectangular Columns. The design charts in this section have been prepared in accordance with the assumption laid down in Part 1 , with the intention that they may be used as standard charts and avoid duplication of effort by individual design offices. ${ }^{3}$

### 2.3 DESIGN SOFTWARE DEVELOPMENT

Since the research of this project is done on design software, it is important to identify the background history of these design software. These software were written by programming software called FORMULA TRANSLATOR (FORTRAN) and C++ Programming Language.

Software engineering revolution began since last 30 years ago. It all begins when FORTRAN was invented. This wonderful first FORTRAN compiler was designed and written from scratch in 1954-57 by an International Business Machine (IBM) team lead by John W. Backus and staffed with super-programmers. However, problems aroused because it was difficult to implement: they were more complicated than traditional finite difference methods, and often the data structures involved are not easily represented in the traditional procedural programming environments used in scientific computing. ${ }^{4}$

In order to solve this problem, a collection of libraries written in a mixture of Fortran and C++ Programming Language were used, In this approach, the high-level data abstractions are implemented in $\mathrm{C}++$, while the bulk of the floating point work is performed on rectangular arrays by Fortran routines. The design approach used here is based on two ideas. The first is that the mathematical structure of the algorithm, domain specified above maps naturally into a combination of data structures and operations on those data structures, which can be embodied in C++ classes. The second is that the mathematical structure of the algorithms can be factored into a hierarchy of abstractions, leading to an analogous factorization of the framework into reusable components, or layers.

Object oriented techniques, and $\mathrm{C}++$ in particular, seem to be taking the software world by storm. Nevertheless, it seems that C++ itself is a major factor in this latest phase of the software revolution. $\mathrm{C}++$ is a programming language suitable for real world projects that is also a more expressive software design language. This results in a more robust design, in essence a better-engineered design. ${ }^{5}$

With the aid from design software, engineers can get through with analysis, design, and detailing in the most convenient method. They are also pretty sure it can be built using accepted construction techniques. Before such a design is actually built the engineers do structural analysis; they build computer models and run simulations; they build scale models and test them.

In short, the software can give designers to make sure the design is a good design before it is built such as:

- Automatic calculation of all building dead loads from structural components.
- Automatic distribution of all uniform and/or concentrated slab loading onto supporting members.
- Automatic creation of necessary analysis models to perform complete building design including automatic pattern loading in accordance with building codes.
- All elements can be designed together in an automated batch design mode. Alternatively, you can interactively control the design of every element or element group
- Layout plans
- All slab reinforcement layouts in plan and/or in section.
- All beam elevation drawings including all reinforcement detailing
- Column Schedules and elevation drawings.
- Complete summary of all analysis output including lateral analysis summaries.
- Complete design calculations for all elements.
- Generation of all material quantities.


### 2.4 DESCRIPTION ON TALL BUILDINGS STRUCTURE. 2.4.1 INTRODUCTION:

Since the research of this project will analyze the RC structures, it will design the structure for tall building later on. Hence it is necessary to understand some criteria in designing this structure.

For the structural engineer the major difference between low and tall buildings is the influence of the wind forces on the behavior of the structural elements. Generally, a tall building structure is one in which the horizontal loads are an important factor in the structural design. In terms of lateral deflections a tall concrete building, which the structure, sized for gravity loads only, will exceed the allowable sway due to additionally applied lateral loads. This allowable drift is set by the code of practice. If the combined horizontal and vertical loads cause excessive bending moments and shear forces the structural system must be augmented by additional bracing elements.

The analysis of tall structures pertains to the determination of the influence of applied loads on forces and deformations in the individual structural elements such as beams, columns and walls. The design deals with the proportioning of these members. For reinforced concrete structures this includes sizing the concrete as well as the steel in an element. Structural analyses are commonly based on established energy principles assume linear elastic behavior of the structural elements. Non-linear behavior of the structure makes the problem extremely complex. It is very difficult to formulate, with reasonable accuracy, the problems involving inelastic responses of building materials.

At present the forces in structural components and the lateral drift of tall structures can be determined by means of elastic method of analysis regardless of the
method of design. Non-linear methods of analysis for high-rise structures are not readily available. ${ }^{6}$

### 2.4.2 DESIGN AND ANALYSIS CONSIDERATIONS.

As stated in of BS 8110: Part 1, clause 2.1, the aim of design is the achievement of an acceptable probability that structures being designed will perform satisfactory during their intended life. F or multi-storey structures the imposed floor loads can be substantially reduced in the design of columns, walls, beams and foundations. Details are given in BS 6399: Part 1, clause 5. BS 8110 contains additional clauses for structures consisting of five storey or more.

## a) Ultimate limit state

i. Structural stability

Tall slender frames may buckle laterally due to loads that are much smaller than predicted by buckling equations applied to isolated columns. Instability may occur for a variety of reasons such as slenderness, excessive axis loads and deformations, cracks, creep, shrinkage, temperature changes and rotation of foundations. Most of these are ignored in a first-order analysis of tall structures but may cause lateral deflections that are much larger than initially expected. The increased deformations can induce substantial additional bending moments in axially loaded members. This will increase the probability of buckling failure. In principle the instability of the multi-storey building structure is no different from that of a low structure but because of the great height of such buildings horizontal deflections must be computed with great accuracy. The deflected shapes of individual structural members should be taken into account in the final analysis of tall slender structures.

## ii. Robustness

All structures should be capable of safely resisting a notional horizontal load applied at each floor or roof level simultaneously. In the design of tall structures it will also be necessary to identify key elements. These can be defined as important structural members whose failure will result in an extended collapse of a large part of the building. ${ }^{5}$

## b) Serviceability limit state

Ideally the limit states of lateral deflection should be concerned with cases where the side sway can
i. limit the use of the structure
ii. influence the behavior of non-load bearing elements
iii. affect the appearance of the structure

## c) Assumptions for analysis

The structural form of a building is inherently three-dimensional. The development of efficient methods of analysis for tall structures is possible only if the usual complex combination of many different types of structural members can be reduced or simplified whilst still representing accurately the overall behavior of the structure. A necessary first step is therefore the selection of an idealized structure that includes only the significant structural elements with their dominant modes of behavior. Achieving a simplified analysis of a large structure such as a tall building is based on two major considerations:
i. the relative importance of individual members contributing to the solution
ii. the relative importance of modes of behavior of the entire structure

The user of a computer program is a simple plane frame or a general finite element program, can usually assign any value to the properties of an element even if these are inconsistent with the actual with the actual size of that member. Several simplifying assumptions are necessary for the analysis of tall building structures subject to lateral loading. The following are the most commonly accepted assumptions.

1. All concrete members behave linearly elastically and so loads and displacements are proportional and the principle of superposition applies. Because of its own weight the structure is subjected to a compressive prestress and pure tension in individual members is not likely to occur;
2. Floor slabs are fully rigid in their own plane. Consequently, all vertical members at any level are subject to the same components of translation and rotation in the horizontal plane. This does not hold for very long narrow buildings and for slabs which have their widths drastically reduced at one or more locations;
3. Contribution from the out-of-plane stiffness of floor slabs and structural bents can be neglected;
4. The individual torsional stiffness of beams, columns and planar walls can be neglected;
5. Additional stiffness effects from masonry walls, fireproofing, cladding and other non-structural elements can be neglected;
6. Deformation due to shear in slender structural members can be neglected;
7. Connections between structural elements in cast-in-situ buildings can be taken as rigid;
8. Concrete structures are elastically stable.

One additional assumption that deserves special attention concerns the calculation of the structural properties of a concrete member. The cross-sectional area and flexural stiffness can be based on the gross concrete sections. This will give acceptable results at service loads but leads to underestimation of the deflections at yielding. In principle the bending stiffness of a structural member reflects the amount of reinforcing steel and takes account of cracked sections, which cause variations in the flexural stiffness along the length of the member. These complications, however, are usually not taken into account in a first-order analysis. ${ }^{6}$

### 2.5 CRITERIA IN DESIGNING TALL BUILDINGS STRUCTURE.

### 2.5.1 INTRODUCTION

A building which height creates different conditions in the design, construction and use than the conditions exist for common buildings of a certain region or period. For the structural engineer; a tall building can be defined as one whose structural system must be modified to make it sufficiently economical to resist lateral forces induce due to wind and earthquakes within the prescribe criteria for:
a.) Strength and stability
b.) Drift
c.) Comfort of occupants

The progression of lateral load resisting schemes from elemental beam and column assemblage towards the notion of an equivalent vertical cantilever is a fundamental to any structural system methodology.

At one end of spectrum there are moment resisting frames, which are efficient for buildings in the range of 20 to 30 stories; at the other end there is the generation of tubular systems were placed with the idea that the application of any particular form is economical only over a limited range of building heights. ${ }^{6}$

### 2.5.2 TALL BUILDINGS STRUCTURE CLASSIFICATION

The $c$ lassification of tall buildings could $b e b$ ased on $c$ ertain engineering and system criteria, which define both the physical as well as the design aspects of the building:
a) Materials: steel, concrete, and composites
b) Gravity load resisting systems: floor framing (beam, slabs), columns, trusses and foundations
c) Lateral load resisting system: walls, frames, trusses diaphragms
d) Type and magnitude of lateral loads: wind, seismic
e) Strength and serviceability requirements: drift, acceleration, ductility

In 1984, a rigorous methodology for cataloguing of tall buildings with respect to their structure systems has been developed. The classification involves four distinct levels of framing oriented divisions:
a) Primary Framing System
b) Bracing Sub-System
c) Floor Framing
d) Configuration and Load Transfer

### 2.5.3 FACTORS AFFECTING GROWTH, HEIGHT, AND STRUCTURAL FORM OF TALL BUILDINGS

The feasibility and desirability of high-rise structures have always depended on:
a) the available materials
b) the level of construction technology
c) the state of development of the services necessary for the use of the building

As a result significant advances have occurred from time to time with the advent of a new material, construction facility, or form of service. The main reasons behind the rapid growth of high-rise buildings were:
a) The socio-economic problems that followed industrialization development
b) Increasing demand for space in growing major cities

Development in the high-rise building design and construction is due to:
a) Different structural systems, which have gradually evolved for residential and office buildings, reflecting their differing functional requirements.
b) Advancements in the major construction materials and other services.
c) Advancement in construction machineries, methods and techniques, particularly pre-cast technology.
d) Development of $4^{\text {th }}$ generation structural software and IT technology, etc.

### 2.6 STRUCTURAL DESIGN CRITERIA AND PHILOSOPHY

The structural design criteria for tall buildings define the following aspects, which control the design:
a) Structural Loading
b) Structural Materials
c) Structural System

### 2.6.1 STRUCTURAL LOADING

The term load refers to any effect that result in a need for some resistive efforts on the part of the structure. There are many sources of loads and many ways in which they can be classified. The principal kinds and sources of loads on building structures are the following:
i) Gravity
ii) Wind
iii) Earthquake
iv) Hydraulic pressure
v) Soil pressure
vi) Thermal Changes
vii) Shrinkage
viii) Vibration
ix) Internal Actions
x) Handling

### 2.6.2 STRUCTURAL MATERIALS

In studying or designing a structure, particular properties of materials are concern. These critical properties may split into:
a) Essential structural properties
b) General properties

Essential structural properties include:
i) Strength
ii) Deformation
iii) Hardness
iv) Fatigue resistance
v) Uniformity of physical structure
vi) Creep, shrinkage, and temperature effects

General properties are:
i) Form
ii) Weight
iii) Fire resistance
iv) Coefficient of thermal expansion
v) Durability
vi) Workability
vii) Appearance
viii) Availability and cost

### 2.6.3 STRUCTURAL SYSTEMS

For selecting a structural systems and optimized design, following are the necessary considerations.
a) Strength and Stability
b) Stiffness and Drift Limitations
c) Human Comfort Criteria

## a) Strength and Stability

For the ultimate limit state, prime design requirement is that the building structure should have adequate strength to resist, and to remain stable under the worst probable load actions that may occur during the lifetime of the building including the period of construction.

## b) Stiffness and Drift Limitations

The provision of adequate stiffness, particular lateral stiffness, is the major consideration in the design of tall building for several important reasons. In terms of serviceability limit state:
i) Deflection must be maintained at a sufficiently low level to allow the proper functioning of non-structural components, such as elevators, doors, etc.
ii) To avoid distress in the structure, to prevent excessive cracking and consequent loss of stiffness, and to avoid any redistribution of load to non-load-bearing partitions, infill, cladding or glazing.
iii) The structure must be sufficiently stiff to prevent dynamic motions to becoming large enough to cause discomfort to occupants, prevent delicate work being undertaken

One p arameter that c an estimate the 1 ateral stiffness of a building is the drift index, d efined as the ratio of maximum deflection at the top of building to the total building height. The control of lateral deflections is particular importance for modern buildings.

## c) Human Comfort Criteria

If a tall flexible structure is subjected to lateral or torsion deflections under the action of wind loads, the resulting oscillatory movements can induce a wide range of responses in the building occupants. It is generally agreed that acceleration is the predominant parameter in determining human response to vibration, but other factors
such as period, amplitude, body orientation, visual and acoustic cues and even past experience can be influential. ${ }^{67}$

### 2.7 RIGID FRAME STRUCTURES

### 2.7.1 INTRODUCTION

Rigid frame high-rise structure comprises parallel arranged bents consisting of columns and beams with moment resistant joints. Resistance to h orizontal loading is provided by the bending resistance of the columns, beams and joints.

### 2.7.1.1 RIGID FRAME BEHAVIOUR

The horizontal stiffness of a rigid frame is governed mainly by the bending resistance of the beams, the columns, and the connections, and, in a tall frame, by the axial rigidity of the columns.

The accumulated horizontal shear above any storey of a rigid frame is resisted by shear in the columns of that storey as shown in Figure 2.1 below.


Figure 2.1 Forces and Deformations caused by external shear

The shear causes by storey-height columns to bend in d ouble c urvature, with point of contraflexure at approximately mid span. These deformations of the columns and beams allow raking of the frame and horizontal deflection in each storey. The overall deflected shape of a rigid frame structure due to raking has a shear configuration with concavity upwind, a maximum inclination near the base, and a minimum inclination at the top. This mode of frame deflection is also called shear mode, and such frames may be framed as shear frames.

The overall moment of the external horizontal shear is resisted in each storey level by the couple resulting from the axial tensile and compressive forces in the columns on opposite sides of the structure as shown in Figure 1.2.


Figure 2.2: Forces and Deformations caused by external moments

The external and shortening of columns cause overall bending and associated displacements of the structure. The contribution of overall bending to the total drift,
however, will usually not exceed $10 \%$ of that raking, except in very tall, slender rigid frames. Therefore the overall deflected shape of a high-rise rigid frame usually has a shear configuration.

### 2.7.2 ANALYSIS OF RIGID FRAME STRUCTURE

As highly redundant structures, rigid frames are designed initially on the basis of approximate analysis, after that a detailed analysis and checks are made. The procedure may typically include the following stages:
i. Estimation of gravity load forces in beams and columns by approximate method.
ii. Preliminary estimate of member sizes based on gravity load forces with arbitrary increase in sizes to allow for horizontal loading.
iii. Approximate allocation of horizontal loading to bents and preliminary analysis of member forces in bents.
iv. Check on drift and adjustment of member sizes if necessary.
v. Check on strength of members for worst combination of gravity and horizontal loading, and adjustment of member sizes if necessary.
vi. Computer analysis of total structure for more accurate check on member strengths and drift, with further adjustment of sizes where required. This stage may include the second-order P- $\Delta$ effects of gravity loading on the member forces and drift.
vii. Detailed design of members and connections.

### 2.7.2.1 APPROXIMATE DETERMINATION OF MEMBER FORCES CAUSED BY GRAVITY LOADING

Since a rigid frame is highly redundant; consequently, an accurate analysis can be made only after the member sizes are assigned. Initially therefore member sizes are decided on the basis of approximate forces estimated either by conservative formulas or by simplified method of analysis that are independent of member properties.

## a) Determination of Beam Forces Using Code recommended Formulas

Code recommended formulas for determining the beam forces can be used upon the following conditions:
i) These are applicable of two or more spans, when the longest span does not exceed the shortest by more than $20 \%$.
ii) The uniformly distributed design live load does not exceed three times the dead load.

### 2.7.2.2 APPROXIMATE ANALYSIS OF MEMBER FORCES CAUSED BY HORIZONTAL LOADING

a) Allocation of Loading Between Bents

A first step in approximate analysis of a rigid frame is to estimate the allocation of the external horizontal force to each bent. The loading will come from Wind Analysis.

## b) Member Force Analysis by Portal Method

The portal method allows an approximate analysis for rigid frames without having to specify member sizes and therefore, it is very useful for a preliminary analysis.

This method is most appropriate to rigid frames that deflect directly by raking. Therefore, it is suitable for structure of moderate slenderness and height, and is commonly recommended as useful structures up to 25 storeys height, and a height to width ratio not greater than 4:1.

It is analogous between a set of single single-bay portal frames and a single storey or multi-bay rigid frames as shown in Figure 2.3a and b.

(a)

(b)

Figure 2.3(a): Separate Portal Analogy (b) Separate Portal Superposed

When each of the separate portals carries a share of the horizontal shear, tension occurs in the windward columns and compression in the leeward columns. If these are superposed to simulate the multi-bay frame, the axial forces of the interior columns are eliminated.

The analysis is based on the following assumptions:
i. Horizontal loading on the frame causes double curvature bending of all the columns and beams, with point of contraflexure at mid height of columns and mid span of the beams.
ii. The horizontal shear at mid storey levels is shared between the columns in proportion to the width of passageway each column support.

The method is used to analyze the whole frame, or just a portion of the frame at a selected level. The analysis of the whole frame considers in turn the equilibrium of separate frame modules, each module consisting of a joint with its column and beam segments extending to the nearest points of contraflexure. The sequence of analyzing the modules is from left to right, starting at the top and working down to the base.

The procedure for a whole frame analysis is as follows:
i. Draw a line diagram of the frame and indicate on it the horizontal shear at each mid-story level.
ii. In each story allocate the shear to the columns in proportion to the aisle widths they support, indicating the values on the diagram.
iii. Starting with the top-left module, compute the maximum moment just below the joint from the product or the column shear and the half-storey height.
iv. Find the girder-end moment just to the right of the joint from the equilibrium of the column and girder moments at the joint. The moment at the other end of the girder is of the same magnitude but corresponds to the opposite curvature.
v. Evaluate the girder shear by dividing the girder end-moment by half the span.
vi. Consider next the equilibrium of the second joint, repeating steps iii to v to find the maximum moment in the second column, and the moment and shear in the second girder from the left. ${ }^{7}$

### 2.8 PROSPECT OF WIND-DRIVEN NATURAL VENTILATION IN TALL BUILDINGS.

## a) Wind Climate of Peninsular Malaysia

The mean surface winds over peninsular Malaysia are generally mild, with the mean speed of about $1.5 \mathrm{~m} / \mathrm{s}$, and a maximum speed of less than $8 \mathrm{~m} / \mathrm{s}$. The main direction is variable. ${ }^{8}$

### 2.9 WIND ANALYSIS

Blowing wind tends to exert loads on buildings and other structures exposed to the wind blowing. The amount of loads induced by wind loading depends on:
a. Wind Speed
b. Building Geometry and Configuration
c. Site Location and Topographical Condition

### 2.9.1 WIND SPEED

a) Basic Wind Speed (V)

According to BS 6399-2, 1997, a basic wind speed is the hourly, mean wind speed at height of 10 m over completely flat terrain at sea level that would occur if the roughness of the terrain was uniform everywhere.

## b) Site Wind Speed ( $\mathbf{V}_{\mathrm{b}}$ )

The basic wind speed modified to account for the altitude of the site and the direction of wind being considered.

## c) Effective (Design) Wind Speed $\left(\mathbf{V}_{s}\right)$

The site wind speed modified to gust speed by taking account of the effective height, the size of the building or structural elements.

### 2.9.2 BACKGROUND

A moving mass of air has kinetic energy; the amount of this energy is directly proportional to the square of the wind velocity:

$$
\mathrm{KE}=\frac{1}{2} \mathrm{mV}^{2}
$$

Where, KE, is the kinetic energy, m , is the wind mass, and V , is the wind velocity. This kinetic energy translates into strain energy when it encounters a stationary object, such as buildings, through deformations induced in that object.

BS Codes presented the following simplified procedure of wind analysis of building structures. The design wind speed, $\mathrm{V}_{\mathrm{b}}$ is converted into dynamic pressure, $\mathrm{q}_{\mathrm{i}}$ at different levels of a building as shown in figure below using the formula:


WIND


Figure 2.4: Wind Calculations on a Multi-storey Frame

$$
q_{i}=0.613 V_{s}^{2}
$$

Where:

$$
\mathrm{q}_{\mathrm{i}}=\text { is dynamic pressure in } \mathrm{KN} / \mathrm{m}^{2}
$$

$\mathrm{V}_{\mathrm{b}}=$ is basic wind speed of a given site in $\mathrm{m} / \mathrm{s}$
$\mathrm{V}_{\mathrm{s}}=$ is design wind speed in $\mathrm{m} / \mathrm{s}$
Where:

$$
V_{s}=S_{1} S_{2} S_{3} V_{b}
$$

Where, $\mathrm{S}_{1} \mathrm{~S}_{2} \mathrm{~S}_{3}$ are given in the table

Wind velocities and pressures


| Struclure |  | Topographatal fictor | Height ofstrature $h$ (mi) |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | 5 | 10 | 15 | 20 | 30 | 40 | 50 | (6) | 81 | 100 | 120 | 140 | 160 | 180 | 20\% |
| Cladding elc. |  |  | 1 | 0.88 | 1,00 | 1.03 | 1.06 | 1,(4) | 1.12 | 1.14 | 1.15 | 1.18 | 1.20 | 1.22 | 1.24 | 1.25 | 1.26 | 8.27 |
|  |  | 2 | 0.79 | 0.93 | 1.00 | 1.03 | 1.07 | 1.10 | 1.12 | 1.14 | 1.17 | 1.19 | 1.21 | 1.22 | 1.24 | 1.25 | 1.26 |
|  |  | 3 | 0.70 | 0.78 | 0.88 | 0.95 | 1.01 | 1.05 | 1.08 | 1.10 | 1.13 | 1.16 | 1.18 | 1.20 | 1.21 | 1.23 | 1.24 |
|  |  | 4 | 0.60 | 0.67 | 0.74 | 0.79 | 0.90 | 0.97 | 1.02 | 1:05 | 1.10 | 1.13 | 1.15 | 1.17 | 1.19 | 1.20 | 1.22 |
|  | + 50 m | I | 0.83 | 0.05 | 0.94 | 1.181 | 1.15 | 1.08 | 1.10 | 1.12 | 1.15 | 1.17 | 1.11 | 1,20 | 1.22 | 1.23 | 1.24 |
|  |  | 2 | 0.74 | 0.ss | 0.95 | 0.98 | 1.10 .1 | 1.0 | 1.08 | 1.11 | 1.1! | 1.16 | 1.18 | 1.19 | 1.21 | 1.72 | 1.24 |
|  |  | 3 | 0.65 | 0.74 | 0.83 | 0.90 | 0.97 | 1.01 | 1.104 | 1.116 | 1,10 | 1.12 | 1.15 | 1.17 | I.1s | 1.20 | 1.21 |
|  |  | 4 | 0.55 | 0.62 | 0.69 | 0.75 | 0.85 | 0.93 | 0.98 | 1.02 | 1.07 | 1.10 | 1.13 | 1.15 | 1.17 | 1.19 | 1.21 |
|  | $>50 \mathrm{~m}$ | I | 0.78 | 0.90 | 0.94 | 0.96 | 1.00 | 1.03 | 1.06 | 1.08 | 1.13 | 1.13 | 1.15 | 1.17 | 1.19 | 1.20 | 1.21 |
|  |  | 2 | 0.70 | 0.83 | 0.91 | 0.94 | 0.98 | 1.01 | 1.04 | 1.06 | 1.09 | 1.12 | 1.14 | 1.16 | 1.18 | 1.19 | 1.21 |
|  |  | 3 | 0.60 | 0.69 | 0.78 | 0.85 | 0.22 | 0.96 | 1.00 | 1.02 | 1.06 | 1.09 | 1.11 | 1.13 | 1.15 | 1.17 | 1.18 |
|  |  | 4 | 0.50 | 0.58 | 0.64 | 0.70 | 0.79 | 0.89 | 0.94 | 0.98 | 1.03 | 1.07 | 1.10 | 1.12 | 1.14 | 1.16 | 1.18 |
| Notes |  |  |  |  |  |  | Topographical fartors |  |  |  |  |  |  |  |  |  |  |
| his height (in metres) ubwe general level of terrain to lop of <br>  On edge of dill or steep hill. |  |  |  |  |  |  | 1. upen cuuntry with no ubstructions <br> 2. opent coundry with scattered wind-breaks <br> 3. country with many wind-breaks; small towns, suburbs of latge citics <br> 4. city eentres and oiher environments with large and frapuent otstructions. |  |  |  |  |  |  |  |  |  |  |

Figure 2.5: Wind velocities tables

## CHAPTER 3 METHODOLOGY

Throughout the project, these steps have be taken to ensure the completion of tasks:

1. Following the examples for each software program to ensure the appropriate ways to operate with the software.
2. Extending the knowledge from Step 1 to solve simple problem. Each problem will give more understanding on how the analysis and design is achieved.
3. Verifying the results obtained from simple structure.
4. Using the software to solve multiple Reinforced Concrete structure problems.
5. Analyzing the results from multiple structures.
6. Discussion and recommendation will be made according to the results.

### 3.1 THE DESIGN PROCESS

Design in any field is a logical creative process, which requires a wide variety of skills. As a complete process, structural engineering design can be divided into three main stages:
a. Conceptual design
b. Preliminary analysis and design
c. Detailed analysis and design

The first stage consists of the drawing up the structural schemes, which are safe, buildable, economical and robust. The second stage consists of performing preliminary calculations to determine if the proposed structural schemes are feasible. Rules of thumb are used to determine preliminary sizes for the various members and approximate methods used to check these sizes and to estimate the quantities of reinforcement required. In the third stage, the adequacy of the preliminary member sizes is verified and the quantities of reinforcement calculated accurately. ${ }^{10}$

Following completion of these stages, drawings and specifications are prepared for the construction of the chosen structure.

### 3.2 ANALYSIS OF FRAMES (MANUAL CALCULATION)

Most concrete buildings contain a structure of beams and columns which, when rigidly connected, make up a continuous frame. The framework of this building concealed behind wall panels which protect the occupants of the building from the external environment.

The analysis of a complete three-dimensional frame can be carried out by hand or by computer using any appropriate method such as the stiffness method. However, the, mathematical complexity of the solution process generally makes it unfeasible to analyze a complete three-dimensional structure by hand. Even when analyzing by computer, the solution may become unduly complex.

One particular aspect of analysis which makes it as yet impractical to design a complete three-dimensional structure is the need to consider all possible arrangements of load. In theory, every possible combination of permanent, variable and wind loading must be considered to determine the critical load effects in each member. The greater the number of members in the frame, the greater the number of possible combinations of applied load. For this reason, certain assumptions and simplifications are commonly made before the structure is analyzed.

In order to overcome the complexity, of considering the full multi-storey skeletal structure and to facilitate frame with smaller, two-dimensional sub-frames. This substantially reduces the total number of load cases which must be considered for each sub-frame and simplifies the process of describing the structural model to the computer. The precise method of simplification depends on whether or not the original frame is braced against horizontal loads. A frame which is braced against horizontal loads using substantial bracing members is termed as non-sway frame.

Owing to the presence of such stiff bracing members, there is little or no lateral deflection in non-sway frame. For this reason, such a frame is designed to resist only the applied vertical loads. A frame that undergoes significant horizontal deflection under applied horizontal loads especially wind load is known as a sway frame. Sway frames must be designed to resist both vertical and horizontal loads. ${ }^{11}$

### 3.2.1 ANALYSIS OF NON-SWAY FRAMES

The first simplification which can be made is to assume that, in the E-W direction, the frame can be represented by three two-dimensional non-sway frames. Note that the vertical loadings for the two outer plan frames are the same and hence only one need to be analyzed. The central plan frame carries a greater vertical load since it supports a greater floor area.


Figure 3.1: Two dimensional Sub-frame

The plane frame can be readily be analyzed by computer for each possible arrangement of load. However, two alternative methods are available for further simplifying the plane frame to facilitate a hand solution.

The first of these methods is to divide the plane frame into a set of sub-frames, each of which is analyzed separately. Each sub-frame is made up of the beams at one level together with the columns connected to these beams. The plane frame can be divided into the three sub-frames below. The columns meeting the beams are assumed to be fixed at their ends. ${ }^{9}$

These sub-frames can readily be analyzed by hand using the moment distribution method to give the moments, shears, etc., in both beams and the columns.
(a)

(b)

(c)


Figure 3.2: Sub-frames for the frame of Figure :(a)top; (b) middle; (c) bottom

## DL

Slab finishes $=0.5 \times 1.7 \mathrm{KN} / \mathrm{m}^{2} \times 3 \mathrm{~m} \quad=2.55 \mathrm{KN} / \mathrm{m}$
Wall Load $\quad=15.12 \mathrm{KN} / \mathrm{m}$
Beam Self-Weight $=0.2 \mathrm{~m} \times 0.45 \mathrm{~m} \times 24 \mathrm{KN} / \mathrm{m}^{3}=2.16 \mathrm{KN} / \mathrm{m}$
Total
$=19.83 \mathrm{KN} / \mathrm{m}$

## $\underline{L L}$

Imposed $=0.5 \times 3 \mathrm{KN} / \mathrm{m}^{2} \times 3 \mathrm{~m} \quad=\underline{4.5 \mathrm{KN} / \mathrm{m}}$

## Ultimate Load

$\mathrm{W}=1.4 \mathrm{DL}+1.6 \mathrm{LL}$
$=34.96 \mathrm{KN} / \mathrm{m}$

### 3.2.1.1 CRITICAL LOADING ARRANGEMENT

For analysis of continuous beam and/or slabs, load is arranged in different manners of load patterns, in order to get the most unfavorable response of the structure. Typical load patterns are shown as:


LOAD PATTERN-1


LOAD PATTERN-2


Figure 3.3: Load Pattern Arrangement

### 3.2.2 ANALYSIS OF SWAY FRAMES

## Wind Analysis



12 m

12 m

Figure 3.4
Height $=$
3.6 m each floor

Figure 3.4 : Plan view of single floor

```
q=0.613Vs\mp@subsup{s}{}{2}
Vs=S1S2S3Vb
S1=S3=1 Vb=8 m/s (maximum wind velocity in Malaysia)
Vs=S2(8m/s)
```

Table 3.1: Wind Load acting on each floor at different height

| $\mathbf{h i}(\mathbf{m})$ | $\mathbf{S} 2$ | $\mathbf{V s}(\mathbf{m} / \mathbf{s})$ | $\mathbf{q} 1\left(\mathbf{N} / \mathbf{m}^{\mathbf{2}}\right)$ | $\mathbf{q 1}\left(\mathbf{K N} / \mathbf{m}^{\mathbf{2}}\right)$ | Point Load(KN) |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 44.00 | 0.950 | 7.60 | 35.41 | 0.0354 | $\mathbf{0 . 1 9 1}$ |
| 40.40 | 0.941 | 7.53 | 34.74 | 0.0347 | $\mathbf{0 . 3 7 5}$ |
| 36.80 | 0.919 | 7.35 | 33.13 | 0.0331 | $\mathbf{0 . 3 5 8}$ |
| 33.20 | 0.890 | 7.12 | 31.08 | 0.0311 | $\mathbf{0 . 3 3 6}$ |
| 29.60 | 0.861 | 6.89 | 29.08 | 0.0291 | $\mathbf{0 . 3 1 4}$ |
| 26.00 | 0.828 | 6.62 | 26.90 | 0.0269 | $\mathbf{0 . 2 9 0}$ |
| 22.40 | 0.792 | 6.34 | 24.61 | 0.0246 | $\mathbf{0 . 2 6 6}$ |
| 18.80 | 0.756 | 6.05 | 22.42 | 0.0224 | $\mathbf{0 . 2 4 2}$ |
| 15.20 | 0.714 | 5.71 | 20.00 | 0.0200 | $\mathbf{0 . 2 1 6}$ |
| 11.60 | 0.668 | 5.34 | 17.51 | 0.0175 | $\mathbf{0 . 1 8 9}$ |
| 8.00 | 0.617 | 4.94 | 14.94 | 0.0149 | $\mathbf{0 . 1 6 1}$ |
| 4.40 | 0.567 | 4.54 | 12.61 | 0.0126 | $\mathbf{0 . 1 5 1}$ |

Wind load per floor:
At typical levels q1 $\times 3.0 \times 3.6$
At the roof level q1 $\times 3.0 \times 1.8$
At the ground level q1 $\times 3.0 \times(1.8+2.2)$
Shear in the top story $=0.191 \mathrm{KN}$

### 3.2.2.1 LATERAL FORCE CALCULATION



### 3.2.2.2 METHODS OF CALCULATIONS

Distributing this shear between the top-story columns in proportion to the widths of aisle supported:

For column A: $0.191 \times 3 / 12=0.048 \mathrm{KN}$
For column B: $0.191(3+3) / 12=0.096 \mathrm{KN}$
For column C: $0.191 \times 3 / 12=0.048 \mathrm{KN}$

The shear in columns of respective stories is allocated.
Moment at top of column $=$ column shear x half-story height

$$
=0.048 \times 1.8=0.0864 \mathrm{KNm}
$$

From moment equilibrium of the joint, the moment at left end of first girder

$$
=-0.0864 \mathrm{KNm}
$$

Shear in girder $=$ girder-end moment/half girder length

$$
=0.0864 / 3=0.029 \mathrm{KN}
$$

Because of the mid-length point of contra flexure, the moment at the right end of the girder has the same value as at the left end. Similarly, the column moments at the top and bottom of a story are equal. The sign convention for numerical values of the bending moment is that an anticlockwise moment a pplied by a joint to the end of a member is taken as positive.

Moment at top of column $=$ column shear x half-story height

$$
=0.096 \times 1.8=0.173 \mathrm{KNm}
$$

From moment equilibrium of the joint, the moment at left end of second girder

$$
=-(0.173-0.0864)=-0.0864 \mathrm{KNm}
$$

Shear in second girder $=$ girder moment/half girder length

$$
=0.0864 / 3=0.029 \mathrm{KN}
$$

## CHAPTER 4 <br> RESULTS AND DISCUSSION

### 4.1 COMPARISON: SOFTWARE RESULTS TO THEORETICAL RESULTS



Figure 4.1: Two dimensional Sub-frame

## FLOOR 1 BEAMS

## MEMBER BF

Table 4.1: Reaction and Moment results for floor 1 beam BF

| Item | Position | Theoretical | Staad Pro | (\%) | Robot | (\%) |
| :---: | :--- | :---: | :---: | :---: | :---: | :---: |
| Reaction (KN) | At left-hand support | 108.34 | 102.23 | 5.6 | - | - |
|  | At right-hand support | -103.25 | -107.54 | 4.2 | -108.48 | 5.1 |
| Moment (KNm) | At left-hand support | 96.47 | 92.26 | 4.4 | - | - |
|  | At right-hand support | 90.58 | 108.20 | 19.5 | 109.03 | 22.4 |

## MEMBERFJ

Table 4.2: Reaction and Moment results for floor 1 beam FJ

| Item | Position | Theoretical | Staad Pro | (\%) | Robot | (\%) |
| :--- | :--- | :---: | :---: | :---: | :---: | :---: |
| Reaction (KN) | At left-hand support | 106.51 | 105.92 | 0.6 | - | - |
|  | At right-hand support | -108.34 | -103.85 | 4.1 | -105.60 | 2.5 |
| Moment (KNm) | At left-hand support | 83.40 | 103.55 | 24.2 | 104.05 | 24.8 |
|  | At right-hand support | 103.65 | 97.32 | 6.1 | - | - |

## FLOOR 1 COLUMNS (4.4m)

Table 4.3: Moment results for floor 1 columns

| Member | Theoretical | Staad Pro | (\%) | Robot | (\%) |
| :---: | :---: | :---: | :---: | :---: | :---: |
| AB | -45.40 | -34.03 | 25.0 | -37.96 | 16.4 |
| BC | 54.66 | 57.87 | 5.9 | 56.86 | 4.0 |
| IJ | -45.40 | -36.38 | 19.9 | -39.69 | 19.9 |
| JK | 54.66 | 60.94 | 11.5 | 59.70 | 9.2 |

## FLOOR 6 BEAMS

## MEMBER CG

Table 4.4: Reaction and Moment results for floor 6 beam CG

| Item | Position | Theoretical <br> $\frac{\text { Typical }}{\text { Floor }^{1}}$ | Staad Pro | (\%) | Robot | (\%) |
| :--- | :--- | :---: | :---: | :---: | :---: | :---: |
| Reaction (KN) | At left-hand support | 109.53 | 99.80 | 8.9 | - | - |
|  | At right-hand support | -104.53 | -109.97 | 5.2 | -109.57 | 4.8 |
| Moment (KNm) | At left-hand support | 66.83 | 70.83 | 6.0 | - | - |
|  | At right-hand support | 59.81 | 101.33 | 69.4 | 102.71 | 71.7 |

[^0]
## MEMBER GK

Table 4.5: Reaction and Moment results for floor 6 beam GK

| Item | Position | Theoretical <br> $\frac{\text { Tvpical }}{\text { Floor }}$ | Staad Pro | (\%) | Robot | (\%) |
| :--- | :--- | :---: | :---: | :---: | :---: | :---: |
| Reaction (KN) | At left-hand support | 105.23 | 108.78 | 3.4 | 105.56 | 0.3 |
|  | At right-hand support | -109.53 | -101.00 | 7.8 | - | - |
| Moment (KNm) | At left-hand support | 59.81 | 98.27 | 64.3 | 99.04 | 65.6 |
|  | At right-hand support | 66.83 | 74.93 | 12.1 | - | - |

## FLOOR 6 COLUMNS

Table 4.6: Moment results for floor 6 columns

| Member | Theoretical | $\underline{\text { Staad Pro }}$ | (\%) | $\underline{\text { Robot }}$ | (\%) |
| :---: | :---: | :---: | :---: | :---: | :---: |
| BC | -33.42 | -32.44 | 2.9 | -32.10 | 3.9 |
| CD | 33.42 | 38.38 | 14.8 | 38.25 | 14.5 |
| JK | -33.42 | -34.69 | 3.8 | -34.79 | 4.1 |
| KL | 33.42 | 40.24 | 20.4 | 40.58 | 21.4 |

## ROOF BEAMS <br> MEMBER DH

Table 4.7: Reaction and Moment results for floor 12 beam DH

| Item | Position | Theoretica! | Staad Pro | (\%) | Robot | (\%) |
| :--- | :--- | :---: | :---: | :---: | :---: | :---: |
| Reaction (KN) | Cantilever support | 6.59 | 6.59 | 0 | 6.66 | 1.1 |
|  | At left-hand support | 20.34 | 28.87 | 41.9 | 29.00 | 42.6 |
|  | At right-hand support | -19.80 | -10.69 | 46.0 | - | - |
|  | Moment (KNm) | Cantilever support | 0.55 | 3.30 | 83.3 | 3.33 |
|  | At left-hand support | 8.43 | 35.04 | 75.9 | 35.15 | 76.0 |
|  | At right-hand support | 8.16 | -19.50 | 58.1 | - | - |

## MEMBER HL

Table 4.8: Reaction and Moment results for floor 12 beam HL

| Item | Position | Theoretical | Staad Pro | (\%) | Robot | (\%) |
| :---: | :--- | :---: | :---: | :---: | :---: | :---: |
| Reaction (KN) | At left-hand support | 19.74 | 10.63 | 46.1 | - | - |
|  | At right-hand support | -20.34 | -28.94 | 42.3 | -29.08 | 43.0 |
|  | Cantilever support | 6.59 | 6.59 | 0 | 6.66 | 1.1 |
| Moment (KNm) | At left-hand support | 8.16 | -19.66 | 58.5 | - | - |
|  | At right-hand support | 8.43 | 35.26 | 76.1 | 35.42 | 78.4 |
|  | Cantilever support | 0.55 | 3.30 | 83.3 | 3.33 | 84.2 |

## ROOF COLUMNS

Table 4.9: Moment results for floor 12 columns

| Member | Theoretical | Staad Pro | (\%) | Robot | (\%) |
| :---: | :---: | :---: | :---: | :---: | :---: |
| CD | -8.43 | -31.74 | 73.4 | -38.55 | 78.1 |
| KL | -8.43 | -31.97 | 73.4 | -38.72 | 78.2 |

### 4.2 COMPARISON: SOFTWARE DETAILING RESULTS

## FLOOR 1 BEAMS

Table 4.10: Reinforcement details for floor 1 beams

| Source | Layers Position |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Top End | Bottom End | Top Middle |  |  | Bottom End | Top End |
|  | Left | Right | Left | Center | Right | Right | Right |
| Staad Pro | 2T20 | 2 T 16 | 2T20 | 3T20 | 2T20 | 2 T 16 | 2T20 |
| Robot Millennium | $\begin{gathered} 2 \mathrm{~T} 12+ \\ 2 \mathrm{~T} 12 \\ \hline \end{gathered}$ | 2 T 16 | 2T20 | $\begin{gathered} 2 \mathrm{~T} 20+ \\ 2 \mathrm{~T} 8 \end{gathered}$ | 2 T 20 | 2 T 16 | 2T20 |

## FLOOR 6 BEAMS

Table 4.11: Reinforcement details for floor 6 beams

| Source | Lavers Position |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Top End | Bottom End | Top Middle |  |  | Bottom End | Top End |
|  | Left | Right | Left | Center | Right | Right | Right |
| Staad Pro | 2T20 | 4 T 12 | 2 T 20 | 3T20 | 2 T 20 | 4 T 12 | 2 T 20 |
| Robot <br> Millennium | $\begin{aligned} & 3 \mathrm{~T} 12+ \\ & 1 \mathrm{~T} 12 \end{aligned}$ | 3T12 | $\begin{gathered} 3 \mathrm{~T} 12+ \\ 2 \mathrm{~T} 8 \\ \hline \end{gathered}$ | $\begin{gathered} 3 \mathrm{~T} 12+ \\ 2 \mathrm{~T} 12 \\ \hline \end{gathered}$ | $\begin{gathered} 3 \mathrm{T12+} \\ 2 \mathrm{~T} 8 \\ \hline \end{gathered}$ | 3 T 12 | $\begin{gathered} 3 \mathrm{~T} 12+ \\ 1 \mathrm{~T} 12 \end{gathered}$ |

## FLOOR 12 BEAMS

Table 4.12: Reinforcement details for floor 12 beams

| Source | Lavers Position |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Top End | $\begin{aligned} & \text { Bottom } \\ & \text { End } \end{aligned}$ | Top Middle |  |  | Bottom End | Top End |
|  | Left | Right | Left | Center | Right | Right | Right |
| Staad Pro |  |  |  | 2 T 12 |  |  |  |
| Robot Millennium | $\begin{gathered} 2 \mathrm{~T} 16+ \\ 2 \mathrm{~T} 16 \end{gathered}$ | 2 T 16 | 2 T 8 | - | 2T8 | 2 T 16 | $\begin{array}{r} 2 \mathrm{~T} 16+ \\ 2 \mathrm{~T} 16 \\ \hline \end{array}$ |

## COLUMNS

Table 4.13: Reinforcement details for columns with different sizes

|  | Size |  |  |
| :---: | :---: | :---: | :---: |
| Source | $250 \mathrm{~mm} \times 250 \mathrm{~mm}$ | 350mmx 250 mm | 400 mmx 400 mm |
| Staad Pro | 8 T 16 |  |  |
| Robot Millennium | 8 T 12 | 6 T 12 | 6 T 12 |

### 4.3 DETAILING BEHAVIOR: LOAD INCREMENT ANALYSIS ON BEAM

## STAAD PRO

Table 4.14: Reinforcement details under different live load imposed using Staad Pro

|  | Layers Position |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Live <br> Load <br> $(\mathbf{K N} / \mathbf{m})$ | Top <br> End | Bottom End | Top Middle |  |  | Bottom End | Top End |
|  | Left | Left | Left | Center | Right | Right | Right |
| $\mathbf{4 . 5 0}$ | 2 T 20 | 2 T 16 | 2 T 20 | 3 T 20 | 2 T 20 | 2 T 16 | 2 T 20 |
| $\mathbf{6 . 0 0}$ | 3 T 20 | 2 T 16 | 2 T 20 | 3 T 20 | 2 T 20 | 2 T 16 | 3 T 20 |
| $\mathbf{7 . 5 0}$ | 3 T 20 | 2 T 16 | 2 T 20 | 3 T 20 | 2 T 20 | 2 T 16 | 3 T 20 |
| $\mathbf{9 . 0 0}$ | 3 T 20 | $2 \mathrm{~T} 12+2 \mathrm{~T} 12$ | 2 T 20 | 3 T 20 | 2 T 20 | $2 \mathrm{~T} 12+2 \mathrm{~T} 12$ | 3 T 20 |
| $\mathbf{1 0 . 5 0}$ | 3 T 20 | $2 \mathrm{~T} 12+2 \mathrm{~T} 12$ | 2 T 20 | 3 T 20 | 2 T 20 | $2 \mathrm{~T} 12+2 \mathrm{~T} 12$ | 3 T 20 |
| $\mathbf{1 2 . 0 0}$ | 2 T 32 | 3 T 16 | 2 T 32 | 2 T 32 | 2 T 32 | 3 T 16 | 2 T 32 |
| $\mathbf{1 3 . 5 0}$ | 2 T 32 | 3 T 16 | 2 T 32 | 2 T 32 | 2 T 32 | 3 T 16 | 2 T 32 |

## ROBOT MILLENNIUM

Table 4.15: Reinforcement details under different live load imposed using Robot Millennium

| Live Load ( $\mathrm{KN} / \mathrm{m}$ ) | Lavers Position |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Top End | Bottom | Top Middle |  |  | Bottom | Top End |
|  | Left | Left | Left | Center | Right | Right | Right |
| 4.50 | $\begin{aligned} & 2 \mathrm{~T} 12+ \\ & 2 \mathrm{~T} 12 \\ & \hline \end{aligned}$ | 2 T 16 | 2 T 20 | $\begin{gathered} \text { 2T20+ } \\ 2 \mathrm{~T} 8 \end{gathered}$ | 2 T 20 | 2 T 16 | 2T20 |
| 6.00 | $\begin{gathered} \hline \text { 3T12+ } \\ 2 \mathrm{~T} 12 \end{gathered}$ | 3 T 12 | $\begin{gathered} \hline \text { 3T12+ } \\ 2 T 8 \end{gathered}$ | $\begin{gathered} 3 \mathrm{~T} 12+ \\ 3 \mathrm{~T} 12 \end{gathered}$ | $\begin{gathered} \text { 3T12+ } \\ 2 \mathrm{~T} 8 \end{gathered}$ | 3 T 12 | $\begin{gathered} 3 \mathrm{~T} 12+ \\ 3 \mathrm{~T} 12 \end{gathered}$ |
| 7.50 | 2T20 | 2 T 12 | $\begin{gathered} \text { 2T16+ } \\ 2 \mathrm{~T} 8 \end{gathered}$ | $\begin{gathered} \text { 2T16+ } \\ 2 \mathrm{~T} 16 \end{gathered}$ | $\begin{gathered} 2 \mathrm{T16+} \\ 2 \mathrm{~T} 8 \end{gathered}$ | 2 T 12 | $\begin{aligned} & 2 \mathrm{T16+} \\ & 2 \mathrm{~T} 16 \\ & \hline \end{aligned}$ |
| 9.00 | 2T20 | 2 T 12 | $\begin{gathered} 2 \mathrm{T16+} \\ 2 \mathrm{~T} 8 \\ \hline \end{gathered}$ | $\begin{gathered} 2 \mathrm{~T} 16+ \\ 2 \mathrm{~T} 16 \\ \hline \end{gathered}$ | $\begin{gathered} 2 \mathrm{~T} 16+ \\ 2 \mathrm{~T} 8 \end{gathered}$ | 2 T 12 | $\begin{aligned} & \hline \text { 2T16+ } \\ & \text { 2T16 } \\ & \hline \end{aligned}$ |
| 10.50 | $\begin{gathered} \hline \text { 3T12+ } \\ 3 \mathrm{~T} 12 \\ \hline \end{gathered}$ | 3 T 12 | $\begin{gathered} 3 T 16+ \\ 2 T 8 \\ \hline \end{gathered}$ | 3 T 20 | $\begin{gathered} \text { 3T16+ } \\ 2 \mathrm{C} 8 \end{gathered}$ | 3 T 12 | $\begin{gathered} 3 \mathrm{T16+} \\ 1 \mathrm{~T} 16 \end{gathered}$ |
| 12.00 | $\begin{gathered} \hline \text { 3T12+ } \\ 3 \mathrm{~T} 12 \\ \hline \end{gathered}$ | 3 T 12 | $\begin{gathered} 3 \mathrm{T16+} \\ 2 \mathrm{~T} 8 \end{gathered}$ | 3 T 20 | $\begin{gathered} \hline \text { 3T16+ } \\ 2 \mathrm{~T} 8 \end{gathered}$ | 3 T 12 | $\begin{gathered} 3 \mathrm{~T} 16+ \\ 1 \mathrm{~T} 16 \end{gathered}$ |
| 13.50 | $\begin{gathered} \text { 3T16+ } \\ \text { 1T16 } \end{gathered}$ | 3 T 12 | $\begin{gathered} 3 \mathrm{~T} 20+ \\ 2 \mathrm{~T} 8 \end{gathered}$ | 3 T 20 | $\begin{gathered} 3 \mathrm{~T} 20+ \\ 2 \mathrm{~T} 8 \end{gathered}$ | 3 T 12 | 3T20 |

### 4.4 DISCUSSION

According to the analysis and design done using Staad Pro, Robot Millennium and manual calculations, there are various different values obtained. It is clearly observed that the results from $m$ anual $c$ alculations $g$ ave a higher value in terms of analysis.

These structural software produced slightly different results, because of different assumptions, specifications, different safety factors and analysis. Staad Pro used matrix displacement method, while Robot Millennium used iterative solver application (Gauss elimination) to avoid factorization of a large-scale matrix. Matrix displacement method used few matrix combinations to obtain the values. The Gauss elimination will solve simultaneous linear equations and simplifying the numerical values.

In the analysis results, the software gave smaller values because the results were analyzed in three-dimensional model while manual calculations only can be done in two-dimensional model. This aspect is very important in the sense that the three-dimensional model allow the beam and column connection joints resist more loads due to the many load distribution to other members. However, in the case of two-dimensional model, fewer members only resisting the loads imposed and can be done by sub-frame calculations. Thus, it will lead to a much higher values.

From Table 4.1 and Table 4.2, the reaction results gave a very small difference up to $5.6 \%$. The interior moment results gave up to $25 \%$ compared with the exterior with only up to $6.1 \%$. This because the interior beam is stressed by members from many directions: upward, downward, leftward and rightward. These effects also affect the columns moment from Table 4.3, showing that longer span gave higher values.

According to Table 4.4 and Table 4.5, the theoretical results in reaction and moment gave smaller values compare with the software results. These theoretical values is determined using sub-frame calculations and without any influence from the upper and lower floors. Table 4.6 shows that the moments from theoretical results are constant, which is relatively not a precise method to obtain the actual results compared with the software.

Furthermore, manual calculations are done only in two-dimensional model by considering the gravity and lateral load only. This because the three-dimensional model using manual calculations are very complicated. The manual calculation using the most conservative way by always considering the larger values and is not the most accurate way to determine the a nalysis and design of s tructure. By considering the higher result values will lead to a larger size of bars in detailing, thus increase the construction cost. Therefore, the design software used to save the time used in designing of structure and also helping in determine the effective construction cost.

The Tables 4.7 to Table 4.9 show that the difference from $40 \%$ to $80 \%$ in terms of reaction and moment. These values were affected by column sizes and connections between them. The column sizes are more dominant at the upper floor, thus leading to a very huge difference. However, both of these software gave approximately similar values.

According to the detailing results, Staad Pro has provided higher bars for beams used in floor 1 (Table 4.10) and 6 (Table 4.11) compared with Robot Millennium. In floor 1 the top center gave $3 \mathrm{~T} 20\left(942 \mathrm{~mm}^{2}\right)$ compared $2 \mathrm{~T} 20+2 \mathrm{~T} 8$ $\left(729 \mathrm{~mm}^{2}\right)$. In floor 6 the top center gave $3 \mathrm{~T} 20\left(942 \mathrm{~mm}^{2}\right)$ compared $3 \mathrm{~T} 12+2 \mathrm{~T} 12$ $\left(565 \mathrm{~mm}^{2}\right)$. However, in the floor 12 (Table 4.12), Robot Millennium gave higher bars used especially at the cantilever portion.

The detailing figures show that Staad Pro split the bars into few parts and not continuous to the end. It shows only the important bars and symmetrically distributed. As for columns from Table 4.13, Staad Pro provides a continuous bar of 8T16 from larger size towards the smaller sizes. The reason is to reduce less tensioning of bars especially at the connection of bars with different sizes of column. Robot Millennium has given exact values to be used which is not practical for construction purposes and need adjustments.

In terms of load increment analysis done on a single beam, it is clearly showed difference of bars used with a gradually increment in live loads for each $3.0 \mathrm{KN} / \mathrm{m}$ increment (Table 4.14 and Table 4.15). From this behavior, is showed that Staad Pro gave larger bars used compared with Robot Millennium. According to Table 4.14, Staad Pro has provided higher bars for beams. In the top center layers gave 3T20 $\left(942 \mathrm{~mm}^{2}\right)$ compared $2 \mathrm{~T} 20+2 \mathrm{~T} 8\left(729 \mathrm{~mm}^{2}\right)$ under live load $4.5 \mathrm{KN} / \mathrm{m}$. The live load is increased $1.5 \mathrm{KN} / \mathrm{m}$ gradually until $13.5 \mathrm{KN} / \mathrm{m}$.

The difference used in Staad Pro (Table 4.14) is relatively constant increment and symmetrical. However, Robot Millennium gave different approach by selecting the most precise method, by using different bars at different layers (Table 4.15), thus making risky and need further adjustment before issued to site construction.

## CHAPTER 5 CONCLUSION AND RECOMMENDATION

As a conclusion, both of the software: STAAD PRO and Robot Millennium have some difference of output in terms of results. Throughout the research, the usage for the STAAD PRO and ROBOT MILLENNIUM has been determined. From the obtained results, the detailing ability and effectiveness of analysis embedded in these software has been identified. These software design can eliminate the tedious manual calculation works and also help the design engineer to appreciate the capability of the software design.

In terms of analysis, the software gave smaller values in terms of shear force and bending moment when compared with theoretical calculations. Small differences have been found from floor 1 and floor 6 . The huge difference at top floor is affected by higher values because column sizes are more dominant. The joint connections between beam and column are supported by lower columns only.

In terms of detailing and load increment analysis effects, Staad Pro gave larger bars used compared with Robot Millennium. Staad Pro detailing gave more adjustment in terms of safety and easy during installation on site. Robot Millennium using the precise method in detailing can gave the designer of the minimum requirement of steel to be used. Both of these software in detailing output need to be adjusted before it is issued to site. These results affect the cost reduction; improve safety procedures, and ease of installations on site.

As a recommendation for future work, the research can be continued on improving the calculation to detailing results. This research can determine the procedure of transferring the calculations into detailing AutoCAD software.

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## APPENDICES

## APPENDIX A <br> MANUAL CALCULATION

(TWO DIMENSIONAL SUB-FRAME METHOD)

## SPAN B-F-J

## Stiffnesses, $\boldsymbol{k}$

## Beam

$$
\mathrm{I}=\frac{\mathrm{bd}^{3}}{12}=\frac{0.2 \times 0.45^{3}}{12}=1.52 \times 10^{-3} \mathrm{~m}^{4}
$$

## Spans BF and JF

$k_{\mathrm{BF}}=k_{\mathrm{JF}}=\frac{1.52 \times 10^{-3}}{6.0}=0.253 \times 10^{-3} \mathrm{~m}^{4}$
$k_{\mathrm{FB}}=k_{\mathrm{FJ}}=\frac{1.52 \times 10^{-3}}{6.0}=0.253 \times 10^{-3} \mathrm{~m}^{4}$

## Columns

$\mathrm{I}=\frac{\mathrm{bd}^{3}}{12}=\frac{0.4 \times 0.4^{3}}{12}=2.13 \times 10^{-3} \mathrm{~m}^{4}$

## Upper

$k_{\mathrm{U}}=\frac{2.13 \times 10^{-3}}{3.6}=0.593 \times 10^{-3} \mathrm{~m}^{4}$

## Lower

$k_{\mathrm{L}}=\frac{2.13 \times 10^{-3}}{4.4}=0.485 \times 10^{-3} \mathrm{~m}^{4}$
$k_{\mathrm{U}}+k_{\mathrm{L}}=(0.593+0.485) \times 10^{-3}=1.08 \times 10^{-3} \mathrm{~m}^{4}$

## Distribution Factors

Joints B and J
$\Sigma k=0.25+1.08=1.33$
D.F.BF $=$ D.F. ${ }^{\mathrm{JF}}$

$$
=0.25 / 1.33=0.19
$$

D.F.COLS $=1.08 / 1.33=0.81$

## Joint F

$\sum k=0.25+0.25+1.08=1.58$
D.F.FB $=$ D.F.FJ

$$
=0.25 / 1.58=0.16
$$

D.F. $\cdot$ COLS $=1.08 / 1.58=0.68$

## Fixed End Moment (F.E.M.)

## For 1.4DL + 1.6LL

$\mathrm{M}_{\mathrm{BF}}=\mathrm{M}_{\mathrm{JF}}=\mathrm{WL}^{2} / 12=34.96(6)^{2} / 12=104.89 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{FB}}=\mathrm{M}_{\mathrm{FJ}}=-\mathrm{WL}^{2} / 12=-34.96(6)^{2} / 12=-104.89 \mathrm{KNm}$

## For 1.0DL

$\mathrm{M}_{\mathrm{BF}}=\mathrm{M}_{\mathrm{JF}}=\mathrm{WL}^{2} / 12=19.83(6)^{2} / 12=59.49 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{FB}}=\mathrm{M}_{\mathrm{FJ}}=-\mathrm{WL}^{2} / 12=-19.83(6)^{2} / 12=-59.49 \mathrm{KNm}$

## First Loading Case

$$
\begin{aligned}
\text { Shear } V_{\mathrm{BF}} & =\underline{\text { Load }} \\
2 & \left(\mathrm{M}_{\mathrm{BF}}-\mathrm{M}_{\mathrm{FB}}\right) \\
\mathrm{L} & \\
& =\underline{209.76}-\frac{(-96.79+86.99)}{6} \\
& =\underline{106.51 \mathrm{KN}} \\
\text { Shear } \mathrm{V}_{\mathrm{FB}} & =\text { Load }-\mathrm{V}_{\mathrm{BF}} \\
& =209.76-106.51 \\
& =\underline{103.25 \mathrm{KN}}
\end{aligned}
$$

$$
\text { Maximum Moment, span } \begin{aligned}
\mathrm{BF} & =\frac{\mathrm{V}_{\mathrm{BF}}^{2}}{2 \mathrm{~W}}+\mathrm{M}_{\mathrm{BF}} \\
& =\frac{106.51^{2}}{2 \times 34.96}-96.79 \\
& =\underline{\mathbf{6 5 . 4 6} \mathbf{K N m}}
\end{aligned}
$$

$$
\text { Distance from } \mathrm{B}, \mathrm{a}_{1}=\frac{\underline{\mathrm{V}_{\underline{B F}}}}{\mathrm{w}}, ~=106.51 / 34.96
$$

## Second Loading Case

$$
\begin{aligned}
\text { Shear } \mathrm{V}_{\mathrm{BF}} & =\frac{\text { Load }}{2}-\frac{\left(\mathrm{M}_{\underline{B F}}-\mathrm{M}_{\underline{\mathrm{FB}}}\right)}{\mathrm{L}} \\
& =\frac{118.98}{2}-\frac{(-60.01+37.27)}{6} \\
& =\underline{63.28 \mathrm{KN}}
\end{aligned}
$$

Shear $V_{F B}=$ Load $-V_{B F}$
$=118.98-63.28$
$=55.70 \mathrm{KN}$
Maximum Moment, span $B F=\frac{V_{B F}}{2 w}+M_{B F}$

$$
\begin{aligned}
& =\frac{63.28^{2}}{2 \times 19.83}-60.01 \\
& =\underline{40.96 \mathrm{KNm}}
\end{aligned}
$$

Distance from B, $a_{1}=\underline{V_{B F}}$

$$
\begin{aligned}
& \mathrm{w} \\
= & 63.28 / 19.83 \\
= & 3.19 \mathrm{~m}
\end{aligned}
$$

Third Loading Case

$$
\begin{aligned}
\text { Shear } V_{\mathrm{BF}} & =\frac{\text { Load }^{2}}{2}-\frac{\left(M_{B F}-M_{\mathrm{FB}}\right)}{L} \\
& =\frac{209.76}{2}-\frac{(-100.06+79.29)}{6} \\
& =\underline{108.34 \mathrm{KN}}
\end{aligned}
$$

$$
\begin{aligned}
\text { Shear } \mathrm{V}_{\mathrm{FB}} & =\text { Load }-\mathrm{V}_{\mathrm{BF}} \\
& =209.76-108.34 \\
& =\underline{101.42 \mathrm{KN}}
\end{aligned}
$$

Maximum Moment, span $B F=\frac{V_{B F}}{2 w}+M_{B F}$

$$
\begin{aligned}
& =\underline{108.34^{2}}-100.06 \\
& 2 \times 34.96 \\
& =\underline{\mathbf{6 7 . 8 1} \mathbf{~ K N m}}
\end{aligned}
$$

Distance from B, $a_{1}=\underset{\underline{W}}{\underline{V_{F}}}$

$$
\begin{aligned}
& =108.34 / 34.96 \\
& =3.10 \mathrm{~m}
\end{aligned}
$$

## First Loading Arrangement

Column Moment
$\mathrm{M}_{\mathrm{BC}}=96.79 \times \frac{0.59}{1.08}=52.88 \mathrm{KNm} \quad \mathrm{M}_{\mathrm{FE}}=124.26 \times \frac{0.49}{1.08}=56.38 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{BA}}=96.79 \times \frac{0.49}{1.08}=43.91 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{JK}}=60.01 \times \frac{0.59}{1.08}=32.78 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{FG}}=124.26 \times \frac{0.59}{1.08}=67.88 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{II}}=60.01 \times \underline{0.49}=27.23 \mathrm{KNm}$
1.08

Second Loading Arrangement
Column Moment

$$
\begin{array}{ll}
\mathrm{M}_{\mathrm{BC}}=60.01 \times \frac{0.59}{1.08}=32.78 \mathrm{KNm} & \mathrm{M}_{\mathrm{FE}}=124.26 \times \frac{0.49}{1.08}=56.38 \mathrm{KNm} \\
\mathrm{M}_{\mathrm{BA}}=60.01 \times \frac{0.49}{1.08}=27.23 \mathrm{KNm} & \mathrm{M}_{\mathrm{JK}}=96.79 \times \frac{0.59}{1.08}=52.88 \mathrm{KNm} \\
\mathrm{M}_{\mathrm{FG}}=124.26 \times \frac{0.59}{1.08}=67.88 \mathrm{KNm} & \mathrm{M}_{\mathrm{JI}}=96.79 \times \frac{0.49}{1.08}=43.91 \mathrm{KNm}
\end{array}
$$

Third Loading Arrangement

## Column Moment

$\mathrm{M}_{\mathrm{BC}}=100.06 \times \underline{0.59}=54.66 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{FE}}=158.58 \times \frac{0.49}{1.08}=71.95 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{BA}}=100.06 \times \frac{0.49}{1.08}=45.40 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{JK}}=100.06 \times \frac{0.59}{1.08}=54.66 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{FG}}=158.58 \times \underline{0.59}=86.63 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{II}}=100.06 \times \frac{0.49}{1.08}=45.40 \mathrm{KNm}$

## Moment and Shear Force Diagram for All Loads Arrangements

(1)

(1)
(2)

(2)

(3)

(3)


## Moment and Shear Force Envelopes

Moment and Shear Force Envelopes are the superimposed moment and shear force diagrams obtained from different possible load arrangements. Critical values from moment's envelopes are selected for design purpose.


Bending Moment Envelope (KNm)

108.34

Shear Force Envelope (KN)


Column Bending Envelope (KNm)

## SPAN C-G-K

## Stiffnesses, $\boldsymbol{k}$

## Beam

$$
\mathrm{I}=\frac{\mathrm{bd}^{3}}{12}=\frac{0.2 \times 0.45^{3}}{12}=1.52 \times 10^{-3} \mathrm{~m}^{4}
$$

## Spans CG and KG

$k_{\mathrm{CG}}=k_{\mathrm{KG}}=\frac{1.52 \times 10^{-3}}{6.0}=0.253 \times 10^{-3} \mathrm{~m}^{4}$
$k_{\mathrm{GC}}=k_{\mathrm{GK}}=\frac{1.52 \times 10^{-3}}{6.0}=0.253 \times 10^{-3} \mathrm{~m}^{4}$

## Columns

$$
\mathrm{I}=\frac{\mathrm{bd}}{12}=\frac{0.25 \times 0.25^{3}}{12}=0.33 \times 10^{-3} \mathrm{~m}^{4}
$$

Upper

$$
k_{\mathrm{U}}=\frac{0.33 \times 10^{-3}}{3.6}=0.09 \times 10^{-3} \mathrm{~m}^{4}
$$

## Lower

$k_{\mathrm{L}}=\frac{0.33 \times 10^{-3}}{3.6}=0.09 \times 10^{-3} \mathrm{~m}^{4}$
$k_{\mathrm{U}}+k_{\mathrm{L}}=(0.09+0.09) \times 10^{-3}=0.18 \times 10^{-3} \mathrm{~m}^{4}$

## Distribution Factors <br> \section*{Joints C and K}

$\Sigma k=0.25+0.18=0.43$
D.F.CG $=$ D.F.KG
$=0.25 / 0.43=0.58$

$$
\text { D.F.COLS }=0.18 / 0.43=0.42
$$

Joint G
$\Sigma k=0.25+0.25+0.18=0.68$
D.F.GC $=$ D.F. ${ }_{\text {GK }}$
$=0.25 / 0.68=0.37$
D.F.COLS $=0.18 / 0.68=0.26$

Fixed End Moment (F.E.M.)
For 1.4DL + 1.6LL
$\mathrm{M}_{\mathrm{CG}}=\mathrm{M}_{\mathrm{KG}}=\mathrm{WL}^{2} / 12=34.96(6)^{2} / 12=104.89 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{GC}}=\mathrm{M}_{\mathrm{GK}}=-\mathrm{WL}^{2} / 12=-34.96(6)^{2} / 12=-104.89 \mathrm{KNm}$

## For 1.0DL

$\mathrm{M}_{\mathrm{CG}}=\mathrm{M}_{\mathrm{KG}}=\mathrm{WL}^{2} / 12=19.83(6)^{2} / 12=59.49 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{GC}}=\mathrm{M}_{\mathrm{GK}}=-\mathrm{WL}^{2} / 12=-19.83(6)^{2} / 12=-59.49 \mathrm{KNm}$

## First Loading Case

Shear $\left.\mathrm{V}_{\mathrm{CG}}=\frac{\text { Load }}{2}-\frac{\left(\mathrm{M}_{\mathrm{CG}}\right.}{\mathrm{L}}-\mathrm{M}_{\mathrm{GC}}\right)$
$=\underline{209.76}-\underline{(-61.90+59.81)}$
$=105.23 \mathrm{KN}$
Shear $\mathrm{V}_{\mathrm{GC}}=$ Load $-\mathrm{V}_{\mathrm{CG}}$
$=209.76-105.23$
$=\underline{104.53 \mathrm{KN}}$
Maximum Moment, span $\mathrm{BF}=\frac{\mathrm{V}_{\mathrm{CG}}}{2{ }^{2}}+\mathrm{M}_{\mathrm{CG}}$

$$
=\frac{105.23^{2}}{2 \times 34.96}-61.90
$$

## $=\underline{96.47 \mathrm{KNm}}$

Distance from $\mathrm{C}, \mathrm{a}_{1}=\underline{\mathrm{V}}_{\mathrm{CG}}$

$$
\begin{aligned}
& \mathrm{w} \\
= & 105.23 / 34.96 \\
= & 3.01 \mathrm{~m}
\end{aligned}
$$

Second Loading Case

$$
\begin{aligned}
\text { Shear } V_{\mathrm{CG}} & =\frac{\text { Load }^{2}-\left(\frac{\left.\mathrm{M}_{\mathrm{CG}}-\mathrm{M}_{\mathrm{GC}}\right)}{\mathrm{L}}\right.}{} \\
& =\frac{118.98}{2}-\frac{(-42.83+1.24)}{6} \\
& =\underline{66.42 \mathrm{KN}} \\
\text { Shear } \mathrm{V}_{\mathrm{GC}} & =\text { Load }-\mathrm{V}_{\mathrm{CG}} \\
& =118.98-66.42 \\
& =\underline{52.56 \mathrm{KN}}
\end{aligned}
$$

Maximum Moment, span $\mathrm{CG}=\frac{\mathrm{V}_{\mathrm{CG}}}{2 \mathrm{w}}+\mathrm{M}_{\mathrm{CG}}$

$$
\begin{aligned}
& =\frac{66.42^{2}}{2 \times 19.83}-42.83 \\
& =\underline{\mathbf{6 8 . 4 1} \mathrm{KNm}}
\end{aligned}
$$

Distance from $\mathrm{C}, \mathrm{a}_{1}=\underline{\mathrm{V}_{\underline{C G}}}$ w

$$
\begin{aligned}
& =66.42 / 19.83 \\
& =3.35 \mathrm{~m}
\end{aligned}
$$

Third Loading Case

$$
\text { Shear } \begin{aligned}
V_{C G} & =\frac{L o a d}{2}-\left(M_{C G}-M_{G C}\right) \\
L & \\
& =\frac{209.76}{2}-\frac{(-66.83+38.95)}{6}
\end{aligned}
$$

$$
\begin{aligned}
& =\underline{109.53 \mathrm{KN}} \\
\text { Shear } \mathrm{V}_{\mathrm{GC}} & =\mathrm{Load}-\mathrm{V}_{\mathrm{CG}} \\
& =209.76-109.53 \\
& =100.23 \mathrm{KN}
\end{aligned}
$$

Maximum Moment, span $C G=\frac{{\underset{V}{C G}}^{2}}{2 \mathrm{~W}}+\mathrm{M}_{\mathrm{CG}}$

$$
\begin{aligned}
& =\underline{109.53^{2}}-66.83 \\
& 2 \times 34.96 \\
& =104.75 \mathrm{KNm}
\end{aligned}
$$

Distance from C, $a_{1}=\underline{\underline{V_{C G}}} \underset{\underline{w}}{ }$

$$
\begin{aligned}
& =109.53 / 34.96 \\
& =3.13 \mathrm{~m}
\end{aligned}
$$

## First Loading Arrangement

Column Moment

$$
\mathrm{M}_{\mathrm{CD}}=\mathrm{M}_{\mathrm{CB}}=61.90 \times \frac{0.09}{0.18}=30.95 \mathrm{KNm}
$$

$$
\mathrm{M}_{\mathrm{GH}}=\mathrm{M}_{\mathrm{GF}}=61.05 \times \underline{0.09}=30.53 \mathrm{KNm}
$$

$$
1.08
$$

$\mathrm{M}_{\mathrm{KL}}=\mathrm{M}_{\mathrm{KJ}}=42.83 \times \frac{0.09}{1.08}=21.42 \mathrm{KNm}$

Second Loading Arrangement
Column Moment

$$
\begin{aligned}
& \mathrm{M}_{\mathrm{CD}}=\mathrm{M}_{\mathrm{CB}}=42.83 \times \frac{0.09}{0.18}=21.42 \mathrm{KNm} \\
& \mathrm{M}_{\mathrm{GH}}=\mathrm{M}_{\mathrm{GF}}=61.05 \times \frac{0.09}{1.08}=30.53 \mathrm{KNm} \\
& \mathrm{M}_{\mathrm{KL}}=\mathrm{M}_{\mathrm{KJ}}=61.90 \times \frac{0.09}{1.08}=30.95 \mathrm{KNm}
\end{aligned}
$$

## Column Moment

$$
\mathrm{M}_{\mathrm{CD}}=\mathrm{M}_{\mathrm{CB}}=66.83 \times \frac{0.09}{0.18}=33.42 \mathrm{KNm}
$$

$\mathrm{M}_{\mathrm{GH}}=\mathrm{M}_{\mathrm{GF}}=77.91 \times \frac{0.09}{1.08}=38.96 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{KL}}=\mathrm{M}_{\mathrm{KJ}}=66.83 \times \frac{0.09}{1.08}=33.42 \mathrm{KNm}$
Moment and Shear Force Diagram for All Loads Arrangements
(1)

(1)

(2)

(2)

109.53
(3)


## Moment and Shear Force Envelopes



Bending Moment Envelope (KNm)
109.53


Shear Force Envelope (KN)


Column Bending Envelope (KNm)


## Stiffnesses, $\boldsymbol{k}$

## Beam

$$
\mathrm{I}=\frac{\mathrm{bd}^{3}}{12}=\frac{0.2 \times 0.45^{3}}{12}=1.52 \times 10^{-3} \mathrm{~m}^{4}
$$

## Spans DH and LH

$k_{\mathrm{DH}}=k_{\mathrm{LH}}=\frac{1.52 \times 10^{-3}}{6.0}=0.253 \times 10^{-3} \mathrm{~m}^{4}$
$k_{\mathrm{HD}}=k_{\mathrm{HL}}=\frac{1.52 \times 10^{-3}}{6.0}=0.253 \times 10^{-3} \mathrm{~m}^{4}$

Columns

$$
\mathrm{I}=\frac{\mathrm{bd}^{3}}{12}=\frac{0.25 \times 0.25^{3}}{12}=0.33 \times 10^{-3} \mathrm{~m}^{4}
$$

Lower

$$
k_{\mathrm{L}}=\frac{0.33 \times 10^{-3}}{3.6}=0.09 \times 10^{-3} \mathrm{~m}^{4}
$$

## Distribution Factors

## Joints D and L

$$
\begin{aligned}
& \Sigma k=0.25+0.09=0.34 \\
& \begin{aligned}
\text { D.F.DH } & =\text { D.F.LH } \\
& =0.25 / 0.34=0.74
\end{aligned}
\end{aligned}
$$

D.F.DX $=$ D.F.LY

$$
=0
$$

D.F. COLS $=0.09 / 0.34=0.26$

## Joint H

$\Sigma k=0.25+0.25+0.09=0.59$
D.F. HD $=$ D.F. HL
$=0.25 / 0.59=0.42$
D.F.coLs $=0.09 / 0.59=0.16$

## Fixed End Moment (F.E.M.)

For 1.4DL + 1.6LL
$\mathrm{M}_{\mathrm{DH}}=\mathrm{M}_{\mathrm{LH}}=\mathrm{WL}^{2} / 12=6.59(6)^{2} / 12=19.78 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{HD}}=\mathrm{M}_{\mathrm{HL}}=-\mathrm{WL}^{2} / 12=-6.59(6)^{2} / 12=-19.78 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{XD}}=\mathrm{M}_{\mathrm{YL}}=0$ (no support)
$\mathrm{M}_{\mathrm{DX}}=\mathrm{M}_{\mathrm{LY}}=-\mathrm{WL}^{2} / 12=-6.59(1)^{2} / 12=-0.55 \mathrm{KNm}$
For 1.0DL
$\mathrm{M}_{\mathrm{DH}}=\mathrm{M}_{\mathrm{LH}}=\mathrm{WL}^{2} / 12=4.71(6)^{2} / 12=14.13 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{HD}}=\mathrm{M}_{\mathrm{HL}}=-\mathrm{WL}^{2} / 12=-4.71(6)^{2} / 12=-14.13 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{XD}}=\mathrm{M}_{\mathrm{YL}}=0$ (no support)
$\mathrm{M}_{\mathrm{DX}}=\mathrm{M}_{\mathrm{LY}}=-\mathrm{WL}^{2} / 12=4.71(1)^{2} / 12=-0.39 \mathrm{KNm}$

## First Loading Case

$$
\begin{aligned}
\text { Shear } V_{D H} & =\frac{\text { Load }^{2}}{2}-\left(M_{D H}-M_{H D}\right) \\
L & \frac{28.26}{2}-\frac{(-6.50+0.43)}{6} \\
& =\underline{15.14 \mathrm{KN}} \\
\text { Shear } V_{H D} & =\text { Load }-V_{\mathrm{DH}} \\
& =28.26-15.14 \\
& =\underline{13.12 \mathrm{KN}}
\end{aligned}
$$

Maximum Moment, span $\mathrm{DH}=\frac{\mathrm{V}_{\mathrm{DH}}}{2 \mathrm{w}}+\mathrm{M}_{\mathrm{DH}}$

$$
\begin{aligned}
& =\underline{15.14^{2}}-6.50 \\
& 2 \times 4.71 \\
& =\underline{\mathbf{1 7 . 8 3} \mathbf{~ K N m}}
\end{aligned}
$$

Distance from D, $a_{1}=\underline{V_{D H}}$

$$
\begin{aligned}
& =17.83 / 4.71 \\
& =3.79 \mathrm{~m}
\end{aligned}
$$

## Second Loading Case

$$
\begin{aligned}
\text { Shear } V_{\mathrm{DH}} & =\frac{\mathrm{Load}}{2}-\frac{\left(\mathrm{M}_{\mathrm{DH}}-\mathrm{M}_{\mathrm{HD}}\right)}{\mathrm{L}} \\
& =\frac{39.54}{2}-\frac{(-7.96+8.16)}{6} \\
& =19.74 \mathrm{KN} \\
\text { Shear } \mathrm{V}_{\mathrm{HD}} & =\mathrm{Load}-\mathrm{V}_{\mathrm{DH}} \\
& =39.54-19.74 \\
& =\underline{19.80 \mathrm{KN}}
\end{aligned}
$$

Maximum Moment, span $D H=\frac{V_{D H}}{2 W^{2}}+M_{D H}$

$$
\begin{aligned}
& =\frac{19.74^{2}}{2 \times 6.59}-7.96 \\
& =\underline{\mathbf{2 1 . 6 1} \mathbf{K N m}}
\end{aligned}
$$

Distance from $D, a_{1}=\underline{V_{D H}}$ w

$$
\begin{aligned}
& =21.61 / 6.59 \\
& =3.28 \mathrm{~m}
\end{aligned}
$$

Third Loading Case

$$
\text { Shear } \begin{aligned}
V_{D H} & =\frac{\text { Load }}{2}-\frac{\left(M_{D H}-M_{H D}\right)}{L} \\
& =\frac{39.54}{2}-\frac{(-8.43+5.01)}{6} \\
& =20.34 \mathrm{KN}
\end{aligned}
$$

$$
\begin{aligned}
& \text { Shear } \mathrm{V}_{\mathrm{HD}}=\text { Load }-\mathrm{V}_{\mathrm{DH}} \\
& =39.54-20.34 \\
& =19.20 \mathrm{KN}
\end{aligned}
$$

Maximum Moment, span $\mathrm{DH}=\frac{\mathrm{V}_{\mathrm{DH}}}{2 \mathrm{w}}+\mathrm{M}_{\mathrm{DH}}$

$$
\begin{aligned}
& =\underline{20.34^{2}}-8.43 \\
& 2 \times 6.59 \\
& =\underline{\mathbf{2 2} .96 \mathbf{K N m}}
\end{aligned}
$$

$$
\text { Distance from D, } \begin{aligned}
\mathrm{a}_{1} & =\underset{\mathrm{V}}{\underline{\mathrm{DH}}} \\
& =22.96 / 6.59 \\
& =3.48 \mathrm{~m}
\end{aligned}
$$

First Loading Arrangement
Column Moment
$\mathrm{M}_{\mathrm{DC}}=6.50 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{HG}}=8.59 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{LK}}=7.96 \mathrm{KNm}$

## Second Loading Arrangement

## Column Moment

$\mathrm{M}_{\mathrm{DC}}=7.96 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{HG}}=8.59 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{LK}}=6.50 \mathrm{KNm}$
Third Loading Arrangement
Column Moment
$\mathrm{M}_{\mathrm{DC}}=8.43 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{HG}}=10.01 \mathrm{KNm}$
$\mathrm{M}_{\mathrm{LK}}=8.43 \mathrm{KNm}$
Moment and Shear Force Diagram for All Loads Arrangements


## Moment and Shear Force Envelopes



Column Bending Envelope (KNm)

## APPENDIX B <br> MANUAL CALCULATION <br> (MOMENT DISTRIBUTION METHOD)

## SPAN BFJ

Moment distribution for first loading case

|  |  |  |  |  | F |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Cols. <br> ( $\sum \mathrm{M}$ ) | BF |  | FB | Cols. <br> (2M) | FJ | JF | Cols. $(\Sigma \mathrm{M})$ |
| D.F.s <br> Load <br> KN | 0.81 | 0.19 | 209.76 | 0.16 | 0.68 | 0.16 | 0.19 | 0.81 |
| F.E.M. |  | 104.89 |  | -104.89 |  | -59.49 | 59.49 |  |
| Bal. | -84.96 | -19.93 | 4 | 26.30 | 111.78 | 26.30 | -11.30 | -48.19 |
| C.O. |  | 13.15 |  | -9.96 |  | -5.65 | 13.15 |  |
| Bal. | -10.65 | -2.50 |  | 2.50 | 10.62 | 2.50 | -2.50 | -10.65 |
| C.O. |  | 1.25 |  | -1.25 |  | -1.25 | 1.25 |  |
| Bal. | -1.01 | -0.24 |  | 0.40 | 1.70 | 0.40 | -0.24 | -1.01 |
| c.o. |  | 0.20 |  | -0.12 |  | -0.12 | 0.20 |  |
| Bal. | -0.16 | -0.04 |  | 0.04 | 0.16 | 0.04 | -0.04 | -0.16 |
| M(KNm) | -96.79 | 96.79 |  | -86.99 | 124.26 | -37.27 | 60.01 | -60.01 |

Moment distribution for second loading case


Moment distribution for third loading case


| Bal. | -0.21 | -0.05 |  | 0.05 | 0.20 | 0.05 | -0.05 |
| :--- | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| M(KNm) | 100.06 | $\mathbf{1 0 0 . 0 6}$ | $\cdot$ | -79.29 | $\mathbf{1 5 8 . 5 8}$ | $\mathbf{- 7 9 . 2 9}$ | -0.21 |

## SPAN CGK

Moment distribution for first loading case


Moment distribution for second loading case

|  |  | C |  |  | G |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Cols. | CG |  | GC | Cols. | GK |  | KG | Cols. |
|  | ( LM ) |  |  |  | ( CM ) |  |  |  | ( M ) |
| D.F.s | 0.42 | 0.58 |  | 0.37 | 0.26 | 0.37 |  | 0.58 | 0.42 |
| Load KN |  |  | 118.98 |  |  |  | 209.76 |  |  |
| F.E.M. |  | 59.49 |  | -59.49 |  | -104.89 |  | 104.89 |  |
| Bal. | 24.99 | -34.50 |  | 60.82 | 42.74 | 60.82 | $\sim$ | -60.84 | -44.05 |
| C.O. |  | 30.41 |  | -17.25 |  | -30.42 |  | 30.41 |  |
| Bal. | 12.77 | -17.64 |  | 17.64 | 12.39 | 17.64 |  | -17.64 | -12.77 |
| c.o. |  | 8.82 |  | -8.82 |  | -8.82 |  | 8.82 |  |
| Bal. | -3.70 | -5.12 |  | 6.53 | 4.59 | 6.53 |  | -5.12 | -3.70 |
| C.O. |  | 3.26 |  | -2.56 |  | -2.56 |  | 3.26 |  |
| Bal. | -1.37 | -1.89 |  | 1.89 | 1.33 | 1.89 |  | -1.89 | -1.37 |
| M(KNm) | $\stackrel{-7}{42.83}$ | 42.83 |  | -1.24 | 61.05 | -59.81 |  | 61.90 | -61.90 |

Moment distribution for third loading case


## SPAN DHL

Moment distribution for first loading case

|  | D |  |  |  | H |  |  |  | L |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | DX | Cols. | DH |  | HD | Cols. | HL | 39.54 | LH | Cols. | LY |
|  |  | ( M ) |  |  |  | ( M ) |  |  |  | ( M ) |  |
| D.F.S | 0 | 0.26 | 0.74 |  | 0.42 | 0.16 | 0.42 |  | 0.74 | 0.26 | 0 |
| Load KN | 28.26 |  |  |  |  |  |  |  |  |  | ! |
| F.E.M. | 0.55 |  | 14.13 |  | -14.13 |  | -19.77 |  | 19.77 |  | 0.39 |
| Bal. | 0 | -3.67 | -10.46 |  | 14.24 | 5.42 | 14.24 |  | -14.63 | -5.14 | 0 |
| c.o. | 0 |  | 7.12 |  | -5.23 |  | -7.31 |  | 7.12 |  | 0 |
| Bal. | 0 | -1.85 | -5.27 |  | 5.27 | 2.01 | 5.27 |  | -5.27 | -1.85 | 0 |
| c.o. | 0 |  | 2.63 |  | -2.63 |  | -2.63 |  | 2.63 |  | 0 |
| Bal. | 0 | -0.68 | -1.95 |  | 2.21 | 0.84 | 2.21 |  | -1.95 | -0.68 | 0 |
| c.o. | 0 |  | 1.11 |  | -0.97 |  | -0.97 |  | 1.11 |  | 0 |
| Bal. | 0 | -0.29 | -0.82 |  | 0.82 | 0.31 | 0.82 |  | -0.82 | -0.29 | 0 |
| M(KNm) | 0.55 | -6.50 | 6.50 |  | -0.43 | 8.59 | -8.16 |  | 7.96 | -7.96 | 0.39 |

Moment distribution for second loading case

|  | D |  |  |  | H |  |  |  | L |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | DX | Cols. | DH |  | HD | Cols. | HL |  | LH | Cols. | 4 |
|  |  | ( M ) |  |  |  | ( M ) |  |  |  | ( $\mathrm{M}_{\text {) }}$ |  |
| D.F.S | 0 | 0.26 | 0.74 |  | 0.42 | 0.16 | 0.42 |  | 0.74 | 0.26 | 0 |
| Load KN |  |  |  | 39.54 |  |  |  | 28.26 |  |  |  |
| F.E.M. | 0.39 |  | 19.77 |  | -19.77 |  | -14.13 |  | 14.13 |  | 0.55 |


| Bal. | 0 | -5.14 | -14.63 | 14.24 | 5.42 | 14.24 | -10.46 | -3.67 | 0 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| c.o. | 0 |  | 7.12 | -7.31 |  | -5.23 | 7.12 |  | 0 |
| Bal. | 0 | -1.85 | -5.27 | 5.27 | 2.01 | 5.27 | -5.27 | -1.85 | 0 |
| c.o. | 0 |  | 2.63 | -2.63 |  | -2.63 | 2.63 |  | 0 |
| Bal. | 0 | -0.68 | -1.95 | 2.21 | 0.84 | 2.21 | -1.95 | -0.68 | 0 |
| c.o. | 0 |  | 1.11 | -0.97 |  | -0.97 | 1.11 |  | 0 |
| Bal. | 0 | -0.29 | -0.82 | 0.82 | 0.31 | 0.82 | -0.82 | -0.29 | 0 |
| M(KNm) | 0.39 | -7.96 | 7.96 | -8.16 | 8.59 | -0.43 | 6.50 | -6.50 | 0.55 |

Moment distribution for third loading


# APPENDIX C <br> STAAD PRO AND ROBOT MILLENNIUM ANALYSIS RESULTS DIAGRAM (BEAMS AND COLUMNS) 

## Shear Force Floor 1



## Bending Moment Floor 1



## Column Moment Floor 1

Lower


## Upper



Shear Force Floor 6


Bending Moment Floor 6


Column Moment Floor 6


## Shear Force Floor 12



## Bending Moment Floor 12



Column Moment Floor 12
Lower


## Bending Moment Floor 1



Bending Moment Floor 6


Bending Moment Floor 12


## Shear Force Floor 1



Shear Force Floor 6


Shear Force Floor 12


## APPENDIX D

STAAD PRO AND ROBOT MILLENNIUM DETAILING RESULTS DIAGRAM
(BEAMS AND COLUMNS)





| Final Year Project <br> Software ficensed tasinow Pantiter (LZ ZO | JobNo | Sheod to 3 | Rwo |
| :---: | :---: | :---: | :---: |
|  | Part |  |  |
| THe FlOOR 12 | Ref |  |  |
|  | By | Dina 17 -Sep-04 Chd |  |
| * | File | Dmatic |  |



| $\rightarrow$ Final Year Project | Sob No | Steet No 1 | Rev |
| :---: | :---: | :---: | :---: |
| - Soltware liconsed mos now Panther 1.2 OG | Part |  |  |
|  | Ret |  |  |
|  | By | Dato 17-Sep-04 Chd |  |
| * | File | Dosami |  |



| Final Year Project | Job No | Sheot No 2 | Rev |
| :---: | :---: | :---: | :---: |
| , - Softwre licensed toSnow Pumther [ZO日] | Part |  |  |
| - THe | Ref |  |  |
| $350 \mathrm{~mm} \times 35 \mathrm{Mm}$ (ounn | By | Duts 17-Sep-04 |  |
| iont | File | Cometic |  |





| $\int_{i}^{3}$ | ${ }_{0}$ | $\underset{\text { xect }}{\underset{\sim}{6}}$ | $\underset{\sigma_{3}}{\frac{8}{3}}$ | \% ${ }^{001}$ |
| :---: | :---: | :---: | :---: | :---: |
| ; | - | ¢ | -m | 8 |
| $i$ | $\begin{aligned} & \mathscr{0} \\ & \stackrel{0}{0} \\ & \text { II } \end{aligned}$ | $\begin{aligned} & \text { 毋 } \\ & \text { \%in } \end{aligned}$ | $\begin{aligned} & \bar{\circ} \\ & \text { II } \end{aligned}$ | $\begin{aligned} & \mathscr{O} \\ & \stackrel{1}{I I} \end{aligned}$ |
| $\frac{v}{v}$ | $\stackrel{\infty}{\sim}$ | $\stackrel{N}{\mathrm{~N}}$ | $\stackrel{\circ}{\mathrm{L}}$ | 剓 |
| $7$ | ( | ( $)$ | (*) | (6) |


|  | Steel $\mathrm{T}=38.4 \mathrm{~kg}$ | $\begin{aligned} & \text { Concrete }=0.558 \mathrm{~m} 3 \\ & \text { Formwork }=6.79 \mathrm{~m} 2 \end{aligned}$ |
| :---: | :---: | :---: |
| Beam7... 8 (taver 1) | Bottom cover 30 cm Side cover 30 cm | Formwork $=6.79 \mathrm{~m} 2$ |
| Section $200 \times 450$ | View scale 1:50 Section scale 1:20 |  |



B-B


A-A


Standard Level
STRUCT 2D



|  | $\stackrel{\circ}{7}$ | 别\| | $\frac{5}{20}$ | ${ }_{8}^{8}$ | [071 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| ल | ค | N | ल | - | 8 |
| $\begin{aligned} & \underset{Y}{\mathrm{O}} \\ & \text { iI } \end{aligned}$ | $\frac{\text { ®o }}{\text { III }}$ | $\begin{aligned} & \text { 유 } \\ & \text { II } \end{aligned}$ | $\begin{aligned} & \text { D } \\ & \text { ָ } \\ & \text { II } \end{aligned}$ | $\begin{aligned} & \text { OR} \\ & \stackrel{0}{0} \\ & \text { II } \end{aligned}$ | $\begin{aligned} & \text { 首 } \end{aligned}$ |
| $\stackrel{N}{N}$ | $\stackrel{\infty}{\sim}$ | $\underset{N}{N}$ | $\stackrel{N}{E}$ | $\underset{N}{N}$ | - |
| $\nabla$ | ( | (a) | ( | (6) | (6) |

[^1]

| 8 | $\stackrel{\stackrel{0}{0}}{\substack{\mid \\ z \varepsilon L}}$ | $\stackrel{8}{7}$ | － | $\stackrel{0}{6}$ |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 강 | ¢ | 앙 | 믹 | $\stackrel{\sim}{0}$ | 今 | 8 |
| $\begin{aligned} & \text { O} \\ & \text { NIN } \end{aligned}$ | $\begin{aligned} & \text { O} \\ & \text { in } \end{aligned}$ | $\stackrel{\circ}{\stackrel{\circ}{\\|}}$ | $\begin{aligned} & \mathbf{O} \\ & \text { Nin } \end{aligned}$ | $\stackrel{\text { © }}{\stackrel{\text { III }}{2}}$ | $\begin{aligned} & \text { 粫 } \\ & \hline \end{aligned}$ | $\stackrel{m}{i n}$ |
| $\stackrel{N}{N}$ | $\frac{N}{N}$ | 䄳 | $\stackrel{\mathrm{N}}{\mathrm{E}}$ | $\underset{\mathbb{N}}{\mathbb{N}}$ | $\stackrel{\stackrel{N}{\mathrm{~N}}}{\stackrel{1}{2}}$ | $\stackrel{\infty}{\text { ¢ }}$ |
| （－） | （ | （ $)$ | （9） | （F） | （ง） | （\％） |

[^2]\mp@subsup{\varphi}{p}{}=2.0

```

2 Beam: Beam7... 8
Number: 1

\subsection*{2.1 Material properties:}


\subsection*{2.2 Geometry:}

2.2.2 Span Position L.supp. L R.supp.
\begin{tabular}{lllll} 
P8_1 Span & 0.40 & 5.60 & 0.40
\end{tabular}

Span length: \(L_{0}=6.00(\mathrm{~m})\)
Section from 0.00 to \(5.60(\mathrm{~m})\) \(200.00 \times 450.00(\mathrm{~mm})\) without left slab without right slab

\subsection*{2.5 Calculation options:}
-C alculations according to
: BS 8110
-P recast beam
: no
- C over
\[
\begin{array}{ll}
\text { bottom } & c=30.00(\mathrm{~mm}) \\
: \text { top } & \text { : side } \mathrm{c1}=30.00(\mathrm{~mm})
\end{array}=30.00(\mathrm{~mm})
\]

\subsection*{2.7 Calculation results:}
\begin{tabular}{llllllll} 
No. Type & State & Span & \(x(m)\) & Value & Capacity & Safety factor \\
1. & \(M\left[k N^{*} m\right]\) & ULS & 1 & 0.40 & -71.21 & -59.37 & 0.83 \\
2. \(M\left[k N^{*} \mathrm{~m}\right]\) & ULS & 2 & 12.00 & -83.75 & -53.67 & 0.64
\end{tabular}

\subsection*{2.7.1 Internal forces in ULS}
\begin{tabular}{lllllll} 
Span & \begin{tabular}{l} 
Mtmax. \\
\(\left(\mathrm{KN}^{*} \mathrm{~m}\right)\)
\end{tabular} & \begin{tabular}{l}
Mtmin. \\
\(\left(\mathrm{kN*} N^{*}\right)\)
\end{tabular} & \begin{tabular}{l}
Ml \\
\(\left(\mathrm{kN} N^{*} \mathrm{~m}\right)\)
\end{tabular} & \begin{tabular}{l}
Mp \\
\(\left(\mathrm{kN} N^{*} \mathrm{~m}\right)\)
\end{tabular} & \begin{tabular}{l}
Q \\
\((\mathrm{kN})\)
\end{tabular} & \begin{tabular}{l} 
Qp \\
\((\mathrm{kN})\)
\end{tabular} \\
P7_1 & 57.19 & 0.00 & -71.21 & -91.30 & 94.31 & -101.48 \\
PB_1 & 56.67 & 0.00 & -79.80 & -83.75 & 97.19 & -98.58
\end{tabular}

\subsection*{2.7.2 Internal forces in SLS}
\begin{tabular}{|c|c|c|c|c|c|c|}
\hline Span & Mtrnax. ( \(\mathrm{kN*}\) m) & Mtmin. ( \(\mathrm{kN}^{*} \mathrm{~m}\) ) & MI ( \(\mathrm{kN}^{\star} \mathrm{m}\) ) & \begin{tabular}{l}
Mp \\
(kN*m)
\end{tabular} & \[
\begin{aligned}
& \mathrm{Q}! \\
& (\mathbf{k N})
\end{aligned}
\] & \[
\mathrm{Qp}_{(\mathrm{kN})}
\] \\
\hline P7_1 & 0.00 & 0.00 & 0.00 & 0.00 & 0.00 & 0.00 \\
\hline P8_1 & 0.00 & 0.00 & 0.00 & 0.00 & 0.00 & 0.00 \\
\hline
\end{tabular}

\subsection*{2.7.4 Required reinforcement area}
\begin{tabular}{lllllll} 
Span & \multicolumn{2}{l}{ Span (mm2) } & \multicolumn{2}{c}{ Left support (mm2) } & \multicolumn{2}{c}{ Right support (mm2) } \\
& bottom & top & bottom & top & bottom & top \\
P7_1 & 347.53 & 0.00 & 0.00 & 440.32 & 0.00 & 579.91 \\
P8_1 & 344.14 & 0.00 & 0.00 & 499.01 & 0.00 & 526.49
\end{tabular}

\subsection*{2.7.5 Deflection and cracking}
\begin{tabular}{ll} 
at(s-t) & - Initial deflection due to total load \\
ap(s-t) & - initial deflection due to long-term load
\end{tabular}
ap(l-1) - long-term deflection due to long-term load
a - total deflection
aal - allowable deflection
\(\mathrm{Cw} \quad\) - width of perpendicular cracks
\begin{tabular}{|c|c|c|c|c|c|c|}
\hline Span & atstr) & ap(s-1) & \(\mathrm{ap}(1-1)\) & a & at & C \\
\hline & (mm) & (mm) & (mm) & (mm) & (mm) & (mm) \\
\hline P7-1 & 0.0000 & 0.0000 & 0.0000 & \(0.0000=(\mathrm{LO} / \mathrm{-})\) & -24.0000 & 0.00 \\
\hline P8_1 & 0.0000 & 0.0000 & 0.0000 & \(0.0000=(\mathrm{LO} /--)\) & -24.0000 & 0.00 \\
\hline
\end{tabular}
2.8 Theoretical results - detailed results:
\begin{tabular}{|c|c|c|c|c|c|c|}
\hline \multirow[t]{2}{*}{2.8 .1} & \multicolumn{6}{|l|}{P7_1 : Span from 0.40 to 6.00 (m)} \\
\hline & ULS & & SLS & & & \\
\hline Abscissa & M max. & \(M \min\). & M max. & Mmin . & A bottom & \\
\hline (m) & ( \(\mathrm{kN}^{*} \mathrm{~m}\) ) & ( \(\mathrm{NN}{ }^{*} \mathrm{~m}\) ) & ( \(\mathrm{kN}^{*} \mathrm{~m}\) ) & ( \(\mathrm{NN}^{*} \mathrm{~m}\) ) & (mm2) & (mm2) \\
\hline 0.40 & 0.00 & -71.21 & 0.00 & 0.00 & 0.00 & 440.32 \\
\hline 0.80 & 0.00 & -34.87 & 0.00 & 0.00 & 0.00 & 206.55 \\
\hline 1.40 & 7.01 & 0.00 & 0.00 & 0.00 & 117.00 & 0.00 \\
\hline 2.00 & 36.33 & 0.00 & 0.00 & 0.00 & 215.43 & 0.00 \\
\hline 2.60 & 53.05 & 0.00 & 0.00 & 0.00 & 320.77 & 0.00 \\
\hline 3.20 & 57.19 & 0.00 & 0.00 & 0.00 & 347.53 & 0.00 \\
\hline 3.80 & 48.75 & 0.00 & 0.00 & 0.00 & 293.24 & 0.00 \\
\hline 4.40 & 27.72 & 0.00 & 0.00 & 0.00 & 162.80 & 0.00 \\
\hline 5.00 & 0.00 & -5.90 & 0.00 & 0.00 & 0.00 & 117.00 \\
\hline 5.60 & 0.00 & -52.09 & 0.00 & 0.00 & 0.00 & 314.70 \\
\hline 6.00 & 0.00 & -91.30 & 0.00 & 0.00 & 0.00 & 579.91 \\
\hline & ULS & SLS & & & & \\
\hline Abscissa & \(Q\) max. & Q max. & Cw & & & \\
\hline (m) & (kN) & (kN) & (mm) & & & \\
\hline 0.40 & 94.31 & 0.00 & 0.00 & & & \\
\hline 0.80 & 80.32 & 0.00 & 0.00 & & & \\
\hline 1.40 & 59.34 & 0.00 & 0.00 & & & \\
\hline 2.00 & 38.37 & 0.00 & 0.00 & & & \\
\hline 2.60 & 17.39 & 0.00 & 0.00 & & & \\
\hline 3.20 & -3.57 & 0.00 & 0.00 & & & \\
\hline 3.80 & -24.55 & 0.00 & 0.00 & & & \\
\hline 4.40 & -45.54 & 0.00 & 0.00 & & & \\
\hline 5.00 & -66.50 & 0.00 & 0.00 & & & \\
\hline 5.60 & -87.48 & 0.00 & 0.00 & & & \\
\hline
\end{tabular}
\begin{tabular}{|c|c|c|c|c|c|c|}
\hline 6.00 & -101.48 & 0.00 & \multicolumn{2}{|l|}{0.00} & & \\
\hline 2.8 .2 & \multicolumn{6}{|l|}{P8_1: Span from 6.40 to 12.00 (m)} \\
\hline & ULS & & SLS & & & \\
\hline Abscissa & \(M_{\text {max }}\). & M min. & M max. & \(M \min\). & A bottom & \\
\hline (m) & ( \(\mathrm{kN}^{*} \mathrm{~m}\) ) & (kN*m) & ( \(\mathrm{NN}^{*} \mathrm{~m}\) ) & ( \(\mathrm{NN}^{*} \mathrm{~m}\) ) & (mm2) & (mm2) \\
\hline 6.40 & 0.00 & -79.80 & 0.00 & 0.00 & 0.00 & 499.01 \\
\hline 6.80 & 0.00 & -42.31 & 0.00 & 0.00 & 0.00 & 252.72 \\
\hline 7.40 & 1.30 & 0.00 & 0.00 & 0.00 & 117.00 & 0.00 \\
\hline 8.00 & 32,35 & 0.00 & 0.00 & 0.00 & 190.97 & 0.00 \\
\hline 8.60 & 50.80 & 0.00 & 0.00 & 0.00 & 306.34 & 0.00 \\
\hline 9.20 & 56.67 & 0.00 & 0.00 & 0.00 & 344.14 & 0.00 \\
\hline 9.80 & 49.96 & 0.00 & 0.00 & 0.00 & 300.93 & 0.00 \\
\hline 10.40 & 30.65 & 0.00 & 0.00 & 0.00 & 180.63 & 0.00 \\
\hline 11.00 & 0.00 & -1.22 & 0.00 & 0.00 & 0.00 & 117.00 \\
\hline 11.60 & 0.00 & -45.71 & 0.00 & 0.00 & 0.00 & 274.01 \\
\hline 12.00 & 0.00 & -83.75 & 0.00 & 0.00 & 0.00 & 526.49 \\
\hline & ULS & SLS & & & & \\
\hline Absclssa & \(Q\) max. & Q max. & Cw & & & \\
\hline (m) & (kN) & \[
(\mathrm{kN})
\] & (mm) & & & \\
\hline 6.40 & 97.19 & 0.00 & 0.00 & & & \\
\hline 6.80 & 83.20 & 0.00 & 0.00 & & & \\
\hline 7.40 & 62.23 & 0.00 & 0.00 & & & \\
\hline 8.00 & 41.25 & 0.00 & 0.00 & & & \\
\hline 8.60 & 20.27 & 0.00 & 0.00 & & & \\
\hline 9.20 & -0.69 & 0.00 & 0.00 & & & \\
\hline 9.80 & -21.68 & 0.00 & 0.00 & & & \\
\hline 10.40 & -42.66 & 0.00 & 0.00 & & & \\
\hline 11.00 & -63.62 & 0.00 & 0.00 & & & \\
\hline 11.60 & -84.61 & 0.00 & 0.00 & & & \\
\hline 12.00 & -98.58 & 0.00 & 0.00 & & & \\
\hline
\end{tabular}

\subsection*{2.9 Reinforcement:}
\begin{tabular}{|c|c|c|c|c|c|c|}
\hline \multicolumn{7}{|l|}{\multirow[t]{2}{*}{\begin{tabular}{l}
Longitudinal reinforcement: \\
- bottom (T)
\end{tabular}}} \\
\hline & & & & & & \\
\hline & 2 & \$16.0 & \(1=6.60\) & from 0.03 & to & 6.49 \\
\hline \multicolumn{7}{|l|}{- assembling (top) ( T )} \\
\hline & 2 & \$8.0 & \(1=3.58\) & from 1.11 & to & 4.69 \\
\hline \multicolumn{7}{|c|}{support (T)} \\
\hline & 2 & \$12.0 & \(1=1.48\) & from 0.03 & to & 1.40 \\
\hline 2 & 2 \$12.0 & \(1=0.94\) & from 0.08 & to 0.91 & & \\
\hline
\end{tabular} Transversal reinforcement: - main (T)
stirrups \(23 \quad \$ 8.0 \quad 1=1.15\)
\(e=5^{*} 0.15+6^{*} 0.30+1^{*} 0.25+1^{*} 0.25+6^{*} 0.30+4^{*} 0.15(\mathrm{~m})\)


\section*{Transversal reinforcement:}
- main (T)
\[
\begin{aligned}
& \text { stirrups } 23 \quad \phi 8.0 \quad \mathrm{I}=1.15 \\
& \mathrm{e}=5^{*} 0.15+6^{*} 0.30+1^{*} 0.25+1^{*} 0.25+6^{*} 0.30+4^{*} 0.15(\mathrm{~m})
\end{aligned}
\]

Project: STRUCT 2D

\section*{Level:}
- Name
: Standard Level
- Reference level :---
- Fire rating \(: 0(\mathrm{~h})\)
- Environment class : mild

\section*{Column: Column60}

Number: 1

\subsection*{2.1 Material properties:}
- C oncrete
: C35
Unit weight \(\quad: 24.00(\mathrm{kN} / \mathrm{m} 3)\)
- L ongitudinal reinforcement :T
\[
\mathrm{f}_{\mathrm{cu}}=35.00(\mathrm{~N} / \mathrm{mm} 2)
\]
- T ransversal reinforcement
: R
\(\mathrm{fy}_{\mathrm{y}}=460.00(\mathrm{~N} / \mathrm{mm} 2)\)
\(\mathrm{fy}=250.00(\mathrm{~N} / \mathrm{mm} 2)\)

\subsection*{2.2 Geometry:}
\begin{tabular}{lll} 
2.2.1 & Rectangular & \(250.00 \times 250.00(\mathrm{~mm})\) \\
2.2.2 & Height: L & \(=3.60(\mathrm{~m})\) \\
2.2.3 & Slab thickness & \(=0.00(\mathrm{~m})\) \\
2.2.4 & Beam height & \(=0.45(\mathrm{~m})\) \\
2.2.5 & Cover & \(=30.00(\mathrm{~mm})\)
\end{tabular}

\subsection*{2.3 Calculation options:}
- C alculations according to : BS 8110
- \(P\) recast column : no
- P re-design : no
- S lenderness taken into account : yes
- \(T\) les :to slab
- \(N\) on-sway structure

\subsection*{2.4 Loads:}
\begin{tabular}{|c|c|c|c|c|c|c|c|c|c|c|}
\hline Case & Nature & Group & \(\mathrm{g}_{\mathrm{f}}\) & N & Myu & Myl & Myi & Mzu & MzI & MzI \\
\hline & & & & (kN) & (kN*m) & ( \(\mathrm{NN}^{*} \mathrm{~m}\) ) & ( \(\mathrm{kN}^{*} \mathrm{~m}\) ) & ( \(\mathrm{KN*}^{*} \mathrm{~m}\) ) & (kN*m) & (kN"m) \\
\hline COMB1 & design & 60 & 1.00 & 40.75 & 35.15 & -39.35 & -15.73 & 0.00 & 0.00 & 0.00 \\
\hline
\end{tabular}

\subsection*{2.5 Calculation results:}

\subsection*{2.5.1 Slenderness analysis}
\begin{tabular}{llll} 
Direction \(Y:\) & \multicolumn{3}{l}{ Non-sway structure } \\
Direction \(Z:\) & \multicolumn{2}{l}{ Non-sway structure } & \\
& \(\mathrm{I}_{0}(\mathrm{~m})\) & \(\mathrm{I}_{\mathrm{e}}(\mathrm{m})\) & b \\
Direction \(\mathrm{Y}:\) & 3.60 & 2.70 & 0.75 \\
Direction \(\mathrm{Z}:\) & 3.60 & 2.70 & 0.75
\end{tabular}
\begin{tabular}{ll} 
ley \(/ \mathrm{h}=10.80\) & Short column (slenderness not taken into account). \\
lez/b \(=10.80\) & Short column (slenderness not taken into account).
\end{tabular}

\subsection*{2.5.2 ULS Analysis}
\begin{tabular}{|c|c|c|c|c|}
\hline \multicolumn{5}{|l|}{Design combination: COMB1 (M2)} \\
\hline - E ccentricity: & & \(e_{y}(\mathrm{~mm})\) & \(\mathrm{e}_{\mathrm{z}}(\mathrm{mm})\) & \multirow{3}{*}{\(\mathrm{Mz}=0.00(\mathrm{kN} * \mathrm{~m})\)} \\
\hline static & \(\mathrm{e}_{0}\) : & 0.00 & -965.97 & \\
\hline total & \(\mathrm{e}_{\text {tot }}\) : & 0.00 & 965.97 & \\
\hline Reinforcement - required area: & & \(\mathrm{A}=896\) & m2) & \\
\hline Ratio: & & \(\mathrm{m}=1\). & & \\
\hline
\end{tabular}

\subsection*{2.6 Reinforcement:}

Main bars (T):
- 8 f12.0 \(\mathrm{I}=3.57\) (m)

\section*{Transversal reinforcement ( R ):}
stirrups: \(\quad 23\) f6.0 \(\quad I=0.83(\mathrm{~m})\)
pins

\title{
APPENDIX F \\ STAAD PRO AND ROBOT MILLENNIUM LIVE LOAD INCREMENT \\ DETAILING DIAGRAM \\ (SELECTED BEAMS ONLY)
}




\begin{tabular}{|c|c|c|c|}
\hline \multirow[t]{2}{*}{\begin{tabular}{l}
Final Year Project \\
Sofware llcensed tosnow Panther [20]
\end{tabular}} & Job No & Sheet No & Rev \\
\hline & \multicolumn{3}{|l|}{Part} \\
\hline - \(\because\) - & \multicolumn{3}{|l|}{Ref} \\
\hline ITVE LOAD INCREMENT PNALYSU \(=1010 . \mathrm{M}\) & By & Dete 17-Sep-04 & Chd \\
\hline & Fila & & Daterlime \\
\hline
\end{tabular}





\begin{tabular}{|c|c|c|c|c|c|c|}
\hline \multirow[t]{2}{*}{} & \multicolumn{2}{|l|}{Final Year Project} & Job No & Sheet No 1 & & \multirow[t]{2}{*}{Rev} \\
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\hline \multirow[t]{2}{*}{LIUE LOAD} & & \multirow[t]{2}{*}{ANAHE/515 \(=120 \mathrm{kN} / \mathrm{m}\)} & \multirow[t]{2}{*}{Ref} & & & \\
\hline & INCREMEOT & & & Date 47-Sep-04 & \multicolumn{2}{|l|}{Chd} \\
\hline & & & File & Dater/ & & \\
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\begin{tabular}{|c|c|c|c|}
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\hline \multirow[t]{3}{*}{,ue lomb increment analusis \(=13.50 \mathrm{caj} / \mathrm{m}\)} & Ref & \multicolumn{2}{|l|}{\multirow[b]{2}{*}{}} \\
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\hline & & \multirow[t]{2}{*}{Steel \(\mathrm{T}=38.4 \mathrm{~kg}\)} & \multirow[t]{2}{*}{\[
\begin{aligned}
\text { Concrete } & =0.558 \mathrm{~m} 3 \\
\text { Formwork } & =6.79 \mathrm{~m} 2
\end{aligned}
\]} \\
\hline \multirow[t]{2}{*}{Standard Level} & \multirow[t]{3}{*}{Beam7... 8 (LIUE \(10980.4 .5 \mathrm{kN} / \mathrm{m})\) Section 200x450} & & \\
\hline & & \begin{tabular}{l}
Bottom cover 30 cm \\
Side cover 30 cm
\end{tabular} & Top cover 30 cm \\
\hline STRUCT 2 D & & View scale 1:50 Section scale 1:20 & \\
\hline
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\footnotetext{

\begin{tabular}{|c|c|c|c|}
\hline \multirow[t]{2}{*}{Standard Level} & \multirow[t]{2}{*}{Beam7． 8 （ \(1, ~\) LIAD－ \(9.0 \mathrm{kN} / \mathrm{m}\) ）} & Steel \(\mathrm{T}=65.8 \mathrm{~kg}\) & \[
\begin{aligned}
& \text { Concrete }=0.558 \mathrm{~m} 3 \\
& \text { Formwork }=6.79 \mathrm{~m} 2 \\
& \hline
\end{aligned}
\] \\
\hline & & Bottom cover 30 cm & Top cover 30cm \\
\hline STRUCT 2 & & Side cover 30 cm & \\
\hline & & Section scale 1：20 & \\
\hline
\end{tabular}

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\begin{tabular}{|c|c|c|}
\hline -2 0 & Steel \(\mathrm{T}=40 \mathrm{~kg}\) & \[
\begin{aligned}
& \text { Concrete }=0.558 \mathrm{m3} \\
& \text { Formwork }=6.79 \mathrm{~m} 2
\end{aligned}
\] \\
\hline Q日! - - 0 (LIVE LOAD \(=13.5 \mathrm{KN} / \mathrm{m}\) ) & \begin{tabular}{l}
Bottom cover 30 cm \\
Side cover 30 cm
\end{tabular} & Top cover 30 cm \\
\hline Section \(200 \times 450\) & \begin{tabular}{l}
View scale 1:50 \\
Section scale 1:20
\end{tabular} & \\
\hline
\end{tabular}
Standard Level
struct 20
\begin{tabular}{|c|c|c|c|c|c|}
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\title{
APPENDIX G \\ GANTT CHART FOR FINAL YEAR PROJECT
}

Actual milestone
Suggested Process



\section*{Actual milestone \\ Suggested Process}```


[^0]:    ${ }^{1}$ Indicates calculation from subframe with 3.6 m height, whereas the software generate the exact floor 6 level.

[^1]:    B-B
    

[^2]:    

    |  |  | Steel T $=44.3 \mathrm{~kg}$ | $\begin{aligned} & \text { Concrete }=0.551 \mathrm{~m} 3 \\ & \text { Formwork }=6.75 \mathrm{~m} 2 \end{aligned}$ |
    | :---: | :---: | :---: | :---: |
    | Standard Level STRUCT 2D | Seam28． $29{ }_{(F L O O R 6)}$ |  |  |
    |  |  | Bottom cover 30 cm Side cover 30 cm | Top cover 30cm |
    |  | Section 200x450 | View scale $1: 50$ Section scale 1：20 |  |

    

    | $\%$ |  | - |
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    | $\stackrel{\text { OO}}{\stackrel{1}{I}}$ | $\begin{aligned} & \text { 끙 } \\ & \text { II } \end{aligned}$ | $\begin{aligned} & \text { ஜீ } \\ & \stackrel{\text { III }}{ } \end{aligned}$ |
    | $\stackrel{\propto}{N}$ | $\stackrel{\circ}{i}$ | $\stackrel{\infty}{\text { ¢ }}$ |
    | $\bar{y}$ | (1) | (6) |

    
    Beam40...64 (FlooR $n)$
    Section $200 \times 450$

    ## Standard Level <br> STRUCT 2D

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    Standard Level
    STRUCT 2D
    

    STAAD PRO AND ROBOT MILLENNIUM DETAILING CALCULATIONS (SELECTED BEAMS AND COLUMNS)

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    & \underline{\text { (200AM 1(M1) }} \\
    & \hline
    \end{aligned}
    $$

    

    SHEAR FORCE DIAGRAM
    

    BENDING MOMENT DIAGRAM

    ## Member M1 Span 1

    Detailed BSB110 Design Requirements
    Section Property. $200 \times 449$

    | Span Length | $=0.000 \mathrm{~m}$ | Rectangular section |  |
    | :--- | :--- | :--- | :--- |
    | Wfidth | $=200 \mathrm{~mm}$ | Depth $\quad=460 \mathrm{~mm}$ |  |
    | $T o p=30 \mathrm{~mm}$ | Bottom $=30 \mathrm{~mm}$ | Side $=30 \mathrm{~mm}$ |  |

    Member My Span 1
    Detailed BSE110 Main Reinforcement
    Hogging: at 0 c000 m from the start of the member

    Moment applied to section
    Effective depth of tension reinforcement
    Depth to compression reinforcement
    Redistribution « 10\%, hence
    $K=\frac{M}{b d \times f^{s}}$
    $K \leq K$ hance campression steel not requined.
    $z=d\left(0.5+\left(0.25 \cdot \frac{K}{0.8}\right)\right) \leq 0.05 d$
    $=381.48 \mathrm{~mm}$
    $A_{* r}=\frac{M}{0.85 f_{z}}$
    Tension Bars provided $\quad=2 T 20$
    Actual area of tension reinforcement $\quad=039.52 \mathrm{~mm}$
    Mnimum area of tension reinfor cement $\quad=0.13$ \% $\quad 3.12 .53$
    hinaximuma area of tension reinforcement $\quad=4 \% \quad 3.120 .1$
    Actual \% of tension reintorcement

    Member M1 Span 1
    Detailed BSB110 Main Reinforcement Cont...
    Min nimum ho rizontal distance betiveen bars $=h_{w, s}+5 \mathrm{~mm} \quad=25 \mathrm{~mm} \quad$ 3.12.11.1
    Smallest actual horizontal space beiveen bars $\quad=84 \mathrm{~mm}$
    Mlaximum spacing of tension bars $=\frac{47000}{f} \leq 300 \quad=105 \pi m \quad 3.12 .11 .2 .4$
    Lamest actual space betrueen tension bars $\quad=84 \mathrm{~mm}$

    Maximum clear distance beiveen beam face and nearest main bar in tension

    $$
    =\text { max tension bar spacing } / 2=83 \mathrm{~mm}
    $$

    3.12.11.2.5

    Actual clear distance belveen beam face and nearest main bar in tension
    $=38 \mathrm{~mm}$

    Actual neutral axis depth of section
    $=90.65 \mathrm{~mm}$
    Morment cap acity of section
    $=08.41 \mathrm{kNm}$
    $\therefore$ OK

    | Sagging: at 3.000 m fom the start of the member |  |  |
    | :---: | :---: | :---: |
    | Norment applied to section | $=57.10 \mathrm{kNm}$ |  |
    | Effective depth of tension reinforcement | d $=404 \mathrm{~mm}$ |  |
    | Depth to compression re inforgement | $\mathrm{d}^{\prime}=54 \mathrm{~mm}$ |  |
    | Redistribution < 10\%, hence | $K=0.150$ | 3.4.4.4 |
    | $K=\frac{M}{\mathrm{DC}^{2} \mathrm{f} f}$ | $=0.050$ |  |
    | $K \leq K$ hence compression steel not required. |  |  |
    | $z=d\left(0.5+\left(0.25 \cdot \frac{K}{0.0}\right)^{\text {a }}\right.$ ( $00.05 d$ | $=380.16 \mathrm{~mm}$ |  |
    | $A_{s y}=\frac{M}{0.95 f_{f}}$ | $=342.84 \mathrm{~mm}^{3}$ |  |
    | Tension Bass provided $\quad=2 T 16$ |  |  |
    | Actual area of tension neinforsement | $=402.12 \mathrm{~mm}$ |  |
    | Minimum area of tension reinforcement | =0.13\% | 3.125 .3 |
    | Maximuma are of tension reinforcement | = $4 \%$ | 3.12.日. 1 |
    | Actual \% of tension reinforcement | $=0.45$ \% |  |
    | Member M1 Span 1 |  |  |
    | Detailed BSB110 Main Reinforcement Cont... |  |  |
    |  | $=25 \mathrm{~mm}$ | 3.12.11.1 |
    | Srullest actual horizontal sp ace between bars | $=02 \mathrm{~mm}$ |  |
    | Maximurn spacing of tension bara $=\frac{47000}{f} \leq 300$ | $=180 \mathrm{~mm}$ | 3.12.11.2.4 |
    | Lamest actual space betueen tension bars | $=02 \mathrm{~mm}$ |  |
    | Maximum clear distance between beam face and nearest main bat in tension |  |  |
    | $=\mathrm{max}$ tension bar spacing $/ 2$ | $=90 \mathrm{~mm}$ | 3.12.11.2.5 |
    | Actual clear distance betrueen beam face and nearest main bar in tension |  |  |
    |  | $=38 \mathrm{~mm}$ |  |
    | Actual neutral axis depth of section | $=01.90 \mathrm{~mm}$ |  |
    | Moment eapacity of section | $=86.09 \mathrm{kNm}$ |  |
    |  | $\therefore \mathrm{OK}$ |  |
    | Hogging: at 0.000 m frome the start of the member |  |  |
    | Moment applied to section | $=108.20 \mathrm{kNm}$ |  |
    | Effeckive depth of tension reinforcement | d $=400 \mathrm{~mm}$ |  |
    | Diepth to compression reinforcement | $\mathrm{d}^{\prime}=54 \mathrm{~mm}$ |  |
    | Redistribution $\leqslant 10 \%$, hence | $k=0.150$ | 3.4.4.4 |
    | $K=\frac{M}{E d i s f}$ | $=0098$ |  |
    | $K \leq K$ hence compression steal not requined. |  |  |
    | $z=d\left(0.5+\left(0.25 \cdot \frac{K}{0.9}\right)\right] \leq 0.85 d$ | $=353.40 \mathrm{~mm}$ |  |
    | $A_{i v}=\frac{M}{0.05 \cdot z}$ | $=060.80 \mathrm{~mm}^{\text {²}}$ |  |
    | Tension Bars provided $\quad=3720$ |  |  |
    | Actual area of terssion reinforcement | $=8.82 .48 \mathrm{~mm}$ |  |
    | Minimum area oftension reinfor cement | $=0.13 \%$ | 3.12.5.3 |
    | Maximum area of fension reimforcement | = 4 \% | 3.120.1 |
    | Actual \% of tension reinforcement | $=1.05 \%$ |  |

    ## Member M1 Span 1

    Detailed ES8110 Main Reinforcement Cont...

    | Minimum harizontal distance bebveen bars $=\mathrm{h}_{\text {sat }}+5 \mathrm{~mm}$ | $=25 \mathrm{~mm}$ | 3.12.11.1 |
    | :---: | :---: | :---: |
    | Smallest actual horizontal sp ace beiveen bars | $=32 \mathrm{~mm}$ |  |
    | Maximum spacing of tension hars $=\frac{47000}{\frac{1}{6}} \leq 300$ | $=207 \mathrm{~mm}$ | 3.12.11.2.4 |
    | Lamest actual space beinuen tension bars | $=32 \mathrm{~mm}$ |  |

    Maximum clear distance beiveen beam face and nearest main bar in tension $=$ max tension bat spacing $/ 2=103 \mathrm{~mm}$
    3.42 .11 .2 .5

    Actual clear distance beiveen beam face and nearest main bar in tension
    $=38 \mathrm{~mm}$

    | Actual neutral axis depth of section | $=145.28 \mathrm{~mm}$ |
    | :--- | :--- |
    | Mioment capacity of section | $=138.64 \mathrm{kNm}$ |
    |  | $\therefore 0 \mathrm{~K}$ |

    Member M1 Span 1
    Detailed BSE110. Span / Effective Depth Check

    | Basic span/effective depth ratio |  | $=28.0$ | 3.4.3 |
    | :---: | :---: | :---: | :---: |
    |  | $=\frac{2 \times 400 \times 342.8}{3 \times 462.1} \times 1$ | $=201.45 \mathrm{Nmmr}^{2}$ | 3.465 |
    | Ntad. factor for tension ift. | $=0.55+\frac{(477-f)}{120(0)+M / b d)} \leq 2.0$ | $=123$ | 3.405 |
    | hiod. factor for compression rit. | $=1+\frac{100 \mathrm{~A}}{\mathrm{bd}} \mathrm{m}$ | $=100$ | 3.48 B |
    | Hence, modified span/ effective d | epth ratio | $=31.83$ |  |
    | Actual span / effective depth ratio |  | = 14.85 | SAFE |

    Member M1 Span 1
    Detailed BS日110 Shear Reinforcement
    High shear zone: 0.001 m to 5.175 m

    $$
    \begin{aligned}
    & \text { Maximum shear force uitthin 20ne, } \mathrm{V} \quad=102.29 \mathrm{kN} \\
    & v=\frac{V}{b d} \quad=1.27 \mathrm{~N} / \mathrm{mmi}^{3} \quad 3.452
    \end{aligned}
    $$

    $$
    \begin{aligned}
    & =0.70 \times 0.78 \times 1.00 \times 11.25=0.05 \mathrm{Nmm}^{2}
    \end{aligned}
    $$

    Member M1 Span 1
    Detailed BSE140 Shear Reinforcement Cont...

    | $\left(v^{\prime}+0.4\right) \leqslant v \leqslant 08.8 \psi^{\prime}$, or 5 Nmm |  | 3.453 |
    | :---: | :---: | :---: |
    | spacing prowided, s. | $=175 \mathrm{~mm}$ |  |
    |  | $=48.68 \mathrm{~mm}$ |  |
    | area of links prowided (278), $A_{\text {er }}$ | $=100.53 \mathrm{~mm}$ |  |
    | distance betueen main barin compression 20 | $\leq 150 \mathrm{~mm}$ | 3.12 |

                                    \(\therefore\) OK
    High shear zone: 5.175 m to 5.009 m
    $\begin{array}{ll}\text { Maximum shear torte nithin zone, } V & =107.54 \mathrm{kN} \\ V=\frac{V}{b d} & =134 \mathrm{~N} / \mathrm{m}_{\mathrm{m}} \quad 3.4 .52\end{array}$
    
    
    $=0.76 \times 1.17 \times 1.00 \times / 1.25=0.75 \mathrm{Nmm}$
    $\left(v_{2}+0.4\right) * v<08 \sqrt{\prime} t$ or $5 \mathrm{Nmm} \quad 3.4 .5 .3$
    spacing provided, $s$. $\quad=300 \mathrm{~mm}$
    
    area of links provided (278). $A_{i k} \quad=100.53 \mathrm{~mm}$
    distance betrveen main bar in compression zone and a restrained bar $\leq 150 \mathrm{~mm} \quad 3.12 .72$
    $\therefore$ OK

    ## $11^{\text {TH }}$ FLOOR

    COLUMN 1(M32)
    (250MM X 250 MM )
    

    Main reinforcement T16
    

    23F8@150mm
    $250 \times 250 \mathrm{~mm}$
    
    
    $\therefore$ area by which steel can increase at laps
    $=404150 \mathrm{~mm}$
    $\therefore \mathrm{OK}$
    Member 32. Detailed BSB110 Shear Reirforcement

    | Spacing provided, 5. | $=150 \mathrm{~mm}$ |  |
    | :---: | :---: | :---: |
    | Midajor axis: |  |  |
    | Shear foree, $V$ | $=19.58 \mathrm{kN}$ |  |
    | Noment, M | $=38.53 \mathrm{kNm}$ |  |
    | $v=\frac{V}{b d}$ | $=0.38 \mathrm{~N} / \mathrm{mm}^{\text {m }}$ | 3.452 |

    cOR $\mathrm{f}_{\mathrm{i}}$ and $5 \mathrm{~N} / m m \mathrm{~m}^{2} \quad \therefore$ dimensions adequate
    

    | $=0.70 \mathrm{~N} / \mathrm{mm}^{\circ}$ | 3.45 .4 |
    | :--- | :--- |
    | $=0.8 \mathrm{~N} / \mathrm{mm}^{\circ}$ | Table 3.8 |
    | $=0.03 \mathrm{~N} / \mathrm{mm}^{2}$ | 3.4 .5 .12 |

    $v^{\prime}=v_{4}+0 . \frac{N}{A} \frac{V h}{M} \quad=0.89+0.0 \times 0.57 \times 0.13$
    $=0.03 \mathrm{~N}_{\mathrm{mm}} \quad 3.4 .5 .12$
    
    Area of links required $=0.4 \mathrm{~b} 5 / 0.86 \mathrm{f}$
    $=03.10 \mathrm{~mm}$
    Area of links provided (2R8), $A_{\text {, }}$
    $=100.53 \mathrm{~mm} \mathrm{~m}^{\prime}$

    Minor axis:
    Shear force, $V \quad=10.52 \mathrm{kN}$
    Mornent, MA
    $v=\frac{v}{b d}$
    $=0.00 \mathrm{kNm}$
    $60.8 \sqrt{ } \mathrm{f}_{5,3}$ and $5 \mathrm{Nmmrr} \quad \therefore$ dimensions adequate
    $v_{0}=0.79\left(\frac{100 A}{b_{i} d}\right)\left(\frac{\Phi D}{d}\right)^{4} I Y=0.70 \times 1.18 \times 1.80 \times 1125$
    $f_{s: x}>25 \quad \therefore v_{i s}=v_{e} \times\left(f_{i n} / 25\right)$ (f finited to 40)
    $v^{\prime}=v_{+}+0 . \frac{N}{A} \frac{V h}{M} \quad=0.80+0.0 \times 0.57 \times 1.00$
    $u \leq v_{s}^{\prime} \quad \therefore$ only nominal links required
    Area of links required $=0.4 \mathrm{~b}, 5,10.85 \mathrm{f}_{\mathrm{m}}$
    Area af links provided (2RB). A
    100 ,

    Member 32 - Detziled BS日110 Shear Reinforcement Cont...

    | Minimum size of link bars | $=0 \mathrm{~mm}$ | 3.12 .7 .1 |
    | :--- | :--- | :--- |
    | Actual size of link bars | $=8 \mathrm{~mm}$ |  |
    | Maximum sp acing of lints | $=1.59 \mathrm{~mm}$ | 3.4 .55 .5 .12 .7 .1 |

    ## 1 Level:

    ```
    -N ame :Standard Level
    -R eference level :---
    -F ire rating :0(h)
    -M aximum cracking :0.30(mm)
    - E nvironment class :moderate
    -C oncrete creeping coefficient: ```

